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Brief Communication

Experimental observation of pulsating instability under acoustic field in downward-propagating flames at large Lewis number



Combustion and Flame

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ABSTRACT

According to previous theory, pulsating propagation in a premixed flame only appears when the reduced Lewis number, $\beta(Le-1)$, is larger than a critical value (Sivashinsky criterion: $4(1 + \sqrt{3}) \approx 11$), where β represents the Zel'dovich number (for general premixed flames, $\beta \approx 10$), which requires Lewis number Le > 2.1. However, few experimental observation have been reported because the critical reduced Lewis number for the onset of pulsating instability is beyond what can be reached in experiments. Furthermore, the coupling with the unavoidable hydrodynamic instability limits the observation of pure pulsating instabilities in flames. Here, we describe a novel method to observe the pulsating instability. We utilize a thermoacoustic field caused by interaction between heat release and acoustic pressure fluctuations of the downward-propagating premixed flames in a tube to enhance conductive heat loss at the tube wall and radiative heat loss at the open end of the tube due to extended flame residence time by diminished flame surface area, i.e., flat flame. The thermoacoustic field allowed pure observation of the pulsating from thermal expansion. By employing this method, we have provided new experimental observations of the pulsating instability for premixed flames. The Lewis number (i.e., $Le \approx 1.86$) was less than the critical value suggested previously.

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1. Introduction

Premixed gas combustion is intrinsically unstable due to (1) the hydrodynamic instability [1] caused by the density difference between unburned and burned mixtures and (2) the diffusivethermal instability [2] caused by the imbalance between the mass and thermal diffusivities of the unburned mixture. In particular, the diffusive-thermal instability can result in different types of instability phenomena which are cellular and pulsating instabilities depending on Lewis number (Le). According to the dispersion relation for the adiabatic flame obtained from a linear stability analysis [2], the cellular instability, which is caused by focusing and diverging mass diffusion at the wrinkled flame, takes place when Le is smaller than a critical value, $Le_c = 1 - 2/\beta$. For a general premixed flame, $\beta \approx 10$, thus $Le_c = 0.8$. The pulsating instability appears when the reduced Lewis number, β (*Le*-1), is larger than the Sivashinsky criterion $(4(1 + \sqrt{3}) \approx 11)$ [2], or Le > 2.1. Joulin and Clavin [3] later showed that excessive heat loss can significantly re-

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duce the critical value of the reduced Lewis number to 6 (Le > 1.6) for the onset of pulsating instability in non-adiabatic flames along with constant density to eliminate hydrodynamic instability.

Concerning the experimental studies, the cellular instability can be easily observed based on Markstein's work [4]. For the pulsating instability, however, the thermo-diffusive analyses based on a onestep model [2,3] cannot be applied directly to real flames since (1) the threshold values of β and *Le* are beyond the range that can be attained in premixed combustion experiments and (2) the hydrodynamic instabilities dominate the thermo-diffusive effects at long wavelength perturbations [5]. Although traveling wave instabilities with similar pulsating motions, such as spiral wave and target patterns, were observed in a few studies [6,7], the dynamic mechanisms were not clear because the pulsating behaviors were combined in these experiments with hydrodynamic instabilities. A clear observation of a pure pulsating instability was difficult to accomplish experimentally.

To solve this problem, we utilized the sustained thermoacoustic field, as a state of nature for a premixed flame propagating in an open-ended tube [8-12]. Since the acoustic field formed by the interaction between heat released by the flame and acoustic fluctuations suppresses the hydrodynamic instability, it enables



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 Table 1

 Summary of the properties of premixed gases and corresponding flame instability regimes.

Mix.	C_3H_8	02	N ₂	Φ	$S_{\rm L}~({\rm cm/s})$	β (Le-1)	Regime
1	0.027	0.169	0.804		17.5	9.82	I
2	0.028	0.175	0.797		20.0	9.72	Ι
3	0.029	0.182	0.789		22.5	9.76	Ι
4	0.030	0.188	0.782		25.0	9.84	II
5	0.031	0.194	0.775		27.5	9.85	IV
6	0.032	0.200	0.768	0.8	30.0	9.78	IV
7	0.033	0.206	0.762		32.5	9.84	IV
8	0.034	0.211	0.755		35.0	9.89	III-2
9	0.035	0.217	0.749		37.5	9.92	III-2
10	0.035	0.222	0.743		40.0	10.03	V
11	0.036	0.227	0.737		42.5	10.17	V
12	0.033	0.135	0.832		7.5	0.73	II
13	0.034	0.142	0.823		10.0	0.63	II
14	0.036	0.149	0.816		12.5	0.56	III-1
15	0.037	0.155	0.808		15.0	0.52	III-1
16	0.039	0.161	0.801	1.2	17.5	0.48	IV
17	0.040	0.166	0.794		20.0	0.45	IV
18	0.041	0.172	0.787		22.5	0.41	IV
19	0.042	0.177	0.781		25.0	0.40	V
20	0.046	0.192	0.761		32.5	0.36	V
21	0.051	0.212	0.737		42.5	0.32	V

observation of the pure pulsating instability. Furthermore, the vibrating flat flame created by the primary acoustic instability strengthens heat losses for several reasons. First, the longer residence time for flame propagation due to the diminished displacement velocity caused by relatively smaller flame surface area of the flat flame than curved shape, enhances the conductive heat loss at the tube wall; meanwhile the radiative heat loss at the open end of the tube is also increased. Second, the dynamic contact of the flame edge with the tube wall increases the conductive heat loss through the back-and-forth motion of combustion gas.

This article reports a clearer observation of the pulsating instability than previous studies [6,7] in the absence of the hydrodynamic instability by applying the primary acoustic field accomplished by the above experimental method at $Le \approx 1.86$.

2. Experiments

The propagation tube (50 mm inner diameter, 711 mm length) was fixed vertically and charged with a premixed gas at atmospheric pressure. The tested premixed gases were propane, oxygen and nitrogen at two fixed equivalence ratios of 0.8 and 1.2 but various laminar burning velocities, to produce reduced Le, β (Le-1), as summarized in Table 1. Here, Lewis number was given by the ratio of the thermal diffusivity of mixture to the mass diffusivity of deficient gas species. When the spark igniter was activated near the upper end of the tube, the top lid was simultaneously opened automatically. The time-dependent downward-propagating flame was recorded by two high-speed cameras. The high-speed cameras captured the side view and the inclined bottom view of the propagating flame. The temporal pressure variation was measured with a dynamic pressure sensor located at the bottom end of the tube. The detailed experimental procedure and mixture properties can be found in Supplementary material.

3. Results and discussion

Figure 1 shows typical flame transition behaviors in a spontaneously generated acoustic field. The numerical values at the left hand side of the picture indicate the distance from the upper end of the tube. Six distinct types of flame transitions were identified with variation of S_L or reduced *Le*, β (*Le*-1):



Fig. 1. The features of flame transitions in acoustic field: (I) Mix. 3, (II) Mix. 13, (III-1) Mix. 15, (IV) Mix. 17 and (V) Mix. 19.

- I: Non-vibrating curved flame
- II: Curved \rightarrow vibrating curved flame
- III-1: Curved \rightarrow vibrating flat flame
- III-2: Curved \rightarrow flat $\rightarrow\,$ pulsating instability with acoustic vibration
- IV: Curved \rightarrow flat \rightarrow corrugated \rightarrow turbulent
- V: Curved \rightarrow corrugated \rightarrow turbulent

The flame behavior of Regime III-2 corresponding to the pulsating instability is not included in Fig. 1. The detail will be shown later in Fig. 2. When S_L was low in lean flames, a regime of nonvibrating flame (Regime I) in which the curved flame propagated without velocity fluctuation was identified. However, this regime was only observed in lean flames, because rich flames have relatively low *Le* ($\Phi = 0.8$; *Le* \approx 1.86 and $\Phi = 1.2$; *Le* \approx 1.04) and these are very sensitive to acoustic fluctuations even with very small $S_{\rm L}$ [12]. When $S_{\rm L}$ was low in rich flames, a regime of acoustic instability (Regime II) was identified in which the flame fronts were still curved, however, non-constant acoustic vibration showed up with small acoustic sound. It was noted that the amplitude of the acoustic pressure fluctuation was not saturated during flame travel to the bottom of the tube according to the pressure measurement. When S_L was increased, vibrating flat flames appeared with relatively amplified acoustic sound (Regime III). This was a representative process of primary acoustic instability and the acoustic pressure could be saturated [8,11]. Also, the primary acoustic field suppressed the initial hydrodynamic instability as observed in Fig. 1. When S_I was further increased, the vibrating flames finally resulted in turbulent motions via corrugated structures with violent acoustic sound (Regime IV). This was a typical process of secondary acoustic instability [13]. When S_L was very large, explosive turbulent motion resulted without the process of flat flames (Regime V). These transition behaviors were in accordance with previous observations [8,11].

It was found that the mean displacement velocity of flat flame before transition to the corrugated structure in Mix. 5 ($S_d = 23.4 \text{ cm/s}$; acoustic) was lower than in Mix. 3 ($S_d = 27.7 \text{ cm/s}$; non-acoustic) although S_L in Mix. 5 ($S_L = 27.5 \text{ cm/s}$) was higher than in Mix. 3 ($S_L = 22.5 \text{ cm/s}$). This is due to the relatively small flame surface area (i.e., flat flame) caused by the primary acoustic field. According to previous study [5], the rate of heat released per

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