

Available online at www.sciencedirect.com



Proceedings of the Combustion Institute

Proceedings of the Combustion Institute 34 (2013) 2691-2697

www.elsevier.com/locate/proci

Limiting conditions for flame spread in fire resistant fabrics

Andres F. Osorio^{a,*}, Carlos Fernandez-Pello^a, David L. Urban^b, Gary A. Ruff^b

^a Department of Mechanical Engineering, University of California, Berkeley, CA 94720, USA ^b NASA John H. Glenn Research Center, Cleveland, OH 44135, USA

Available online 3 August 2012

Abstract

Fire resistant (FR) fabrics are used for astronauts, firefighter and racecar driver suits. However, their fire resistant characteristics depend on the environmental conditions and require study. Particularly important is the response of these fabrics to varied environments and radiant heat from a source such as an adjacent fire. In this work, experiments were conducted to study the effect of oxygen concentration, external radiant flux and oxidizer flow velocity on the concurrent flame spread over two FR fabrics: Nomex HT90-40 and a Nomex/Nylon/Cotton fabric blend. Results show that for a given fabric the minimum oxygen concentration for concurrent flame spread depends strongly on the magnitude of the external radiant flux. At increased oxygen concentrations the external radiant flux required for flame spread decreases. Oxidizer flow velocity influences the external radiant flux only when the convective heat flux from the flame has similar values to the external radiant flux. The results of this work provide further understanding of the flammability characteristics of fire resistant fabrics in environments similar to those of future spacecrafts. © 2012 The Combustion Institute. Published by Elsevier Inc. All rights reserved.

Keywords: Fire-resistant; Flame spread; Limiting oxygen index; Charring

1. Introduction

Potential future space exploration missions by NASA or other international space agencies together with commercial space exploration spurred by lower operational costs have the potential to encourage the design of a new generation of spacecrafts and increase the number of humans traveling to space. An increased human presence in space also poses many challenges, particularly in human safety. Fire in a spacecraft environment represents a particularly dangerous situation given the combined effects of extended periods of time, confined space, low-pressure, elevated oxygen concentrations, low flow velocities and micro, or partial, gravity. Ignition and burning of fabrics and other combustible materials during a spacecraft mission could compromise mission success, but most importantly astronaut safety. The determination of the flammability characteristics of fabrics of both nonfire resistant (NFR) and fire resistant (FR) is important for fire safety purposes.

^{*} Corresponding author. Address: Department of Mechanical Engineering, University of California, Berkeley Mailstop 1740, Berkeley, CA 94720, USA. Fax: +1 510 642 1850.

E-mail addresses: andres.osorio@berkeley.edu (A.F. Osorio), ferpello@me.berkeley.edu (C. Fernandez-Pello), david.urban@nasa.gov (D.L. Urban), gary.a.ruff@nasa.gov (G.A. Ruff),

^{1540-7489/\$ -} see front matter © 2012 The Combustion Institute. Published by Elsevier Inc. All rights reserved. http://dx.doi.org/10.1016/j.proci.2012.07.053

Nomenclature

ΔH_c	heat of combustion	1.	burn length
e) time and temperature dependent	l_b	flame length
$\Delta \Pi_p(1, i)$		l_f	
• //	heat of pyrolysis	l_h	heated length
m"	mass loss rate	l_p	pyrolysis length
Q'	heat release rate per unit length	S	solid thickness
\dot{q}_{ext}''	external radiant flux	t	time
$\dot{\dot{Q}'}_{\dot{q}''_{ext}}$ \dot{q}''_{fr}	radiant flux from the flame to the	t_b	gasification time
	solid	T_f	flame temperature
\dot{q}_{MAX}''	maximum external radiant flux at	$\check{T_o}$	solid initial temperature
	sample leading edge	T_p	pyrolysis temperature
$\dot{q}_{MIN}^{\prime\prime}$	minimum heat flux for flame spared	$\dot{U_{\infty}}$	oxidizer flow velocity
\dot{q}_{sr}''	reradiation from the solid	V_f	flame spread rate
$ ho_g$	gas phase density		
ρ_s	solid density	Abbreviations	
С	generic constant	LOI	limiting oxygen index
c_g	specific heat of the gas phase	ULOI	upward limiting oxygen index
c_s	specific heat of the solid	MOC	maximum oxygen concentration
g	gravity	FR	fire resistant
\tilde{k}_g	gas phase thermal conductivity	NFR	non-fire resistant
-			

Several works conducted in normal and reduced gravity have characterized the ignition and flame spread characteristics of many thin materials under varying conditions of material thickness, external heat flux, oxygen concentration, pressure and forced flow velocity. Particularly relevant for the present work are the studies of [1–3] regarding the limiting oxygen concentrations, flammability and flame spread characteristics of FR fabrics. The limiting oxygen index (LOI) and maximum oxygen concentration (MOC) are two oxygen indices used to measure the flammability of a material. In its most common definition, the LOI is the lowest oxygen concentration that supports a flame [1,4]. In other applications, the LOI and MOC have been defined in terms of NASA's upward propagation test NASA-STD-6001 Test 1 [5]. Using this test, the LOI is defined as the oxygen concentration at which a material passes NASA-STD-6001 Test 1 approximately half the time. The MOC is defined as the maximum oxygen concentration at which a material passes NASA-STD-6001 Test 1 [2,3]. Using upward and downward flame spread tests with Nomex III in 1 atm Klenihenz and T'ien [1] determined the LOI to be 24% for upward flame spread, and 28% for downward flame spread. Upward propagation LOI tests are also known was (ULOI). Using reduced pressure environments, Hirsch et al. [2] found that when pressure was reduced to 0.7 atm the ULOI for single layers of Nomex HT90-40 was less than 30%. And in concurrent flammability experiments using single layers of Nomex HT90-40, Olson et al. [3] measured

the MOC to be 22% in 1 and 0 g, while the ULOI was determined to 25.4% in 1 g and 23% in 0 g.

In the present work we present experiments aimed to determine the effects of an external radiant flux, oxidizer flow velocity and oxygen concentration on the minimum conditions for concurrent flame spread over thin flame retarding fabrics. Concurrent flame spread is analogous to upward flame spread, with the only difference being that in the latter the forced flow velocity is replaced with a buoyant velocity. In concurrent flame spread the flame covers the solid fuel during the heating and pyrolysis process, which makes it faster and more hazardous than opposed flame spread. For this reason, the concurrent flame spread configuration was selected for the present experiments. Two FR fabrics, Nomex HT90-40 and a Nomex/Nylon/Cotton fabric blend were used for the experiments. Both are materials are used in astronaut space suits, and other earth fabrics such as firefighter clothing and race car driver suits. Nomex is a family of aromatic polyamide fibers that create a strong flexible polymer chain with a high degree of heat resistance. In normal atmospheric conditions Nomex does not melt or drip but chars when exposed to high temperatures [6], which contributes to its FR characteristics.

2. Apparatus configuration

The experiments are conducted in the apparatus shown schematically in Fig. 1. It consists of a small-scale combustion wind tunnel 560 mm Download English Version:

https://daneshyari.com/en/article/240901

Download Persian Version:

https://daneshyari.com/article/240901

Daneshyari.com