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Frequency analysis of pressure oscillations in large length-to-diameter two-phase micro-channel heat sinks



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ABSTRACT

Two-phase micro-channel heat sinks are prime candidates for incorporation into thermal control systems (TCSs) of future space vehicles and planetary habitats. Unlike small heat sinks employed in the electronics industry, TCS heat sinks are characterized by large length-to-diameter ratio, for which limited information is presently available. This study employs a 609.6-mm long by 203.2-mm wide heat sink containing 100 of 1 x 1 mm² micro-channels and uses R134a as working fluid. The large length-todiameter ratio of 609.6 is especially instrumental to capturing detailed axial variations of flow pattern and corresponding variations in local heat transfer coefficient. High-speed video analysis of the inlet plenum shows appreciable vapor backflow under certain operating conditions, which is also reflected in periodic oscillations in the measured pressure drop. In fact, the backflow frequency captured by video matches closely the frequency obtained from Fourier analysis of the pressure drop signal. While density-wave oscillations are encountered in individual channels, the phenomena observed are more closely related to parallel-channel instability. It is shown the periodic oscillations and vapor backflow are responsible for initiating intermittent dryout and appreciable drop in local heat transfer coefficient in the downstream regions of the channels. A parametric study of oscillation frequency shows a dependence on four dimensionless parameters that account for amount of vapor generation, subcooling, and upstream liquid length, in addition to Weber number. A new correlation for oscillation frequency is constructed that captures the frequency variations relative to these individual parameters.

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1. Introduction

1.1. Emergence of Two-phase cooling technologies

During the past four decades, the rate of heat dissipation in a number of technologies, especially electronics and power applications, has increased enormously, exasperated by a quest for smaller and more lightweight system architectures. To cope with these challenges, a variety of two-phase high-heat-flux thermal management techniques, with device heat fluxes in the range of 10^2-10^3 W/cm², have been proposed, and some have evolved into practical solutions [1]. The most basic of these techniques are passive cooling schemes such as capillary-driven devices (heat pipes, capillary pumped loops, and loop heat pipes) [2–4] and pool boiling thermosyphons [5–7]. But the more demanding cooling situations are tackled with the aid of a mechanical pump to capitalize upon the heat transfer merits of faster fluid motion. The pumped

E-mail address: mudawar@ecn.purdue.edu (I. Mudawar). URL: https://engineering.purdue.edu/BTPFL (I. Mudawar). cooling schemes are the focus of extensive studies that have been performed at the Purdue University Boiling and Two-Phase Flow Laboratory (PU-BTPFL). They include falling film [8], channel flow boiling [9–13], mini/micro-channel flow boiling [14–20], jet-impingement [21–23], and spray [24–28], as well as hybrid cooling schemes combining the merits of mini/micro-channel flow boiling and jet impingement [29].

1.2. Two-phase mini/micro-channel cooling

As discussed in [30], mini/micro-channel flow boiling has received unprecedented attention because of its ability to not only achieve high heat fluxes, but also reduce size and weight of cooling hardware as well as coolant inventory, let alone their design simplicity and low manufacturing cost. It is important to note that single-phase mini/micro-channel heat sinks have also received significant attention, especially in the electronics industry. The primary merit of this single-phase cooling scheme is the inverse dependence of heat transfer coefficient on hydraulic diameter for laminar flow typically encountered in mini/micro-channels, meaning cooling performance may be enhanced simply by decreasing

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Nomenclature cross-sectional area of micro-channel [m²] W_{w} half-width of copper sidewall separating micro- A_{base} total base area of heat sink [m²] channels [m] thermodynamic equilibrium quality boiling number Во χ_e C_n complex Fourier coefficient sampling point in time, $0 \le x_k \le 2\pi$ [rad] χ_k coordinate along micro-channel [m] specific heat at constant pressure [J/kg K] c_p \dot{D}_h hydraulic diameter [m] frequency [Hz] Greek symbols f_k complex form of Fourier series thermal diffusivity [m²/s] marginal frequency [Hz] f_{margin} film thickness [um] discrete Fourier transform fin efficiency η ĥ* complex conjugate of discrete Fourier transform density [kg/m³] Ğ mass velocity [kg/m² s] surface tension [N/m] σ h enthalpy [J/kg]; heat transfer coefficient [W/m² K] τ period [s] height of micro-channel's cross-section [m] H_{ch} latent heat of vaporization []/kg] h_{fg} Subscripts $I_{sp,f}$ dimensionless single-phase liquid length parameter average avg superficial velocity [m/s] bottom of micro-channel k thermal conductivity [W/m K] evap evaporation L micro-channel length [m] experimental ехр heated length of channel [m] L_H liquid fin parameter m liquid-only fo total mass flow rate of heat sink [kg/s] m vapor g Ν number of measurement points in micro-channel inlet; inlet N_{ch} number of micro-channels in heat sink micro-channel outlet: outlet out dimensionless phase change parameter N_{pch} pred predicted N_{sub} dimensionless subcooling parameter solid material of heat sink pressure [Pa] sat saturation Ре Peclet number single-phase SD P_F friction perimeter of channel [m] superheated super heated perimeter of channel [m] P_H thermocouple tc P_r reduced pressure upstream plenum ир pressure drop [Pa] Δp wall w q''heat flux [W/m²] q_B'' heat flux based on total base area of heat sink [W/m²] Acronyms heat flux based on heated perimeter of micro-channel q_H'' CHF critical heat flux $[W/m^2]$ DNB departure from nucleate boiling volumetric flow rate [m³/s] Q DWO density-wave oscillation vapor generation rate [m³/s] Q_g **EBO** explosive boiling oscillation T temperature [°C] LFD Ledinegg instability time; oscillation period [s] ONB onset of nucleate boiling T_{sat} saturation temperature [°C] PCI parallel-channel instability wall temperature [°C] T_w **SPDO** severe pressure-drop oscillation mean fluid velocity [m/s] īı width of micro-channel's cross-section [m] W_{ch} We Weber number

the channel diameter. However, the cooling performance of these single-phase devices is limited by their sole dependence on sensible heat rise of the coolant. On the other hand, two-phase heat sinks can achieve orders of magnitude enhancement in the heat transfer coefficient by utilizing the coolant's both sensible and latent heat. They also result in smaller temperature gradients both between the device and coolant, and axially along the channel.

Despite these merits, two-phase mini/micro-channel heat sinks are not without shortcomings. By decreasing the hydraulic diameter in order to increase the two-phase heat transfer coefficient, pressure drop across the channels can increase appreciably, resulting in significant compressibility (resulting from axial changes of vapor and liquid specific volumes with pressure) and flashing (resulting from axial changes of vapor and liquid enthalpies with pressure), as well as increased likelihood of two-phase choking [30]. These performance drawbacks are, in fact, closely related to the distinction between channel sizes: one to a few millimeters

for mini-channels versus ten to several hundred micrometers for micro-channels. For example, Bowers and Mudawar [31] compared the cooling performances of a mini-channel heat sink with $D_h = 2.5$ mm to that of a micro-channel heat sink with $D_h = 0.51$ mm, and showed that, while they produce fairly similar heat transfer performances, the latter produced significantly higher pressure drop and incurred appreciable compressibility and flashing.

Recently, the authors of the present study explored the implementation of two-phase mini/micro-channel heat sinks as multi-kilowatt evaporators for future space vehicles that could tackle heat removal from both avionics and crew [32]. They also proposed the concept of using the same evaporators in a 'Hybrid Thermal Control System' (H-TCS) for space vehicles, allowing the cooling loop to be automatically reconfigured into a vapor compression loop, a pumped two-phase cooling loop, or a pumped single-phase loop.

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