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Aerodynamic performance investigation on waverider with variable blunt radius in hypersonic flows

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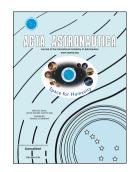
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1	Aerodynamic performance investigation on waverider with
2	variable blunt radius inhypersonic flows
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5	Republic of China
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7	Abstract: Waverider is an important candidate for the design of hypersonic vehicles. However, the ideal
8	waverider cannot be manufactured because of its sharp leading edge, so the leading edge should be blunted. In
9	the paper, the HMB solver and laminar flow model have been utilized to obtain the flow field properties around
10	the blunt waverider with the freestream Mach number being 8.0, and several novel strategies have been
11	suggested to improve the aerodynamic performance of blunt waverider. The numerical method has been
12	validated against experimental data, and the Stanton number (St) of the predicted result has been analyzed. The
13	obtained results show good agreement with the experimental data. St_{max} decreases by 58% and L/D decreases by
14	8.2% when the blunt radius increases from 0.0002m to 0.001m. "Variableblunt waverider" is a good compromise
15	for aerodynamic performance and thermal insulation. The aero-heating characteristics are very sensitive to R_{max} .
16	The position of the smallest blunt radius has a great effect on the aerodynamic performance. In addition, the type
17	of blunt leading edge has a great effect on the aero-heating characteristics when R_{max} is fixed. Therefore, out of
18	several designs, Type 4is the best way to achieve the good overall performance. The "Variable blunt waverider"

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