## Accepted Manuscript

Lateral entry guidance with no-fly zone constraint

Zixuan Liang, Siyuan Liu, Qingdong Li, Zhang Ren

 PII:
 S1270-9638(16)30946-4

 DOI:
 http://dx.doi.org/10.1016/j.ast.2016.10.025

 Reference:
 AESCTE 3811

To appear in: Aerospace Science and Technology

Received date:28 February 2016Revised date:13 September 2016Accepted date:26 October 2016



Please cite this article in press as: Z. Liang et al., Lateral entry guidance with no-fly zone constraint, Aerosp. Sci. Technol. (2016), http://dx.doi.org/10.1016/j.ast.2016.10.025

This is a PDF file of an unedited manuscript that has been accepted for publication. As a service to our customers we are providing this early version of the manuscript. The manuscript will undergo copyediting, typesetting, and review of the resulting proof before it is published in its final form. Please note that during the production process errors may be discovered which could affect the content, and all legal disclaimers that apply to the journal pertain.

## ACCEPTED MANUSCRIPT

1

2

3

### Lateral Entry Guidance with No-Fly Zone Constraint

Zixuan Liang, Siyuan Liu, Qingdong Li<sup>\*</sup>, Zhang Ren

Science and Technology on Aircraft Control Laboratory, Beihang University, Beijing 100191, China

#### 4 Abstract:

5 No-fly zones are geographic constraints for the atmospheric entry flight. Two lateral guidance methods are presented for the entry flight constrained by multiple no-fly zones. The first method 6 7 employs a dynamic heading corridor to control the vehicle's velocity heading angle. The corridor is 8 generated based on a chain mode strategy that takes heading limits for all the no-fly zones and the 9 target into account. The dynamic corridor is adaptively updated during the flight according to the vehicle's state and the actual constraints. The second method is designed based on a waypoint vector 10 11 which is located between each pair of no-fly zones. This waypoint vector contains both the location 12 and the direction constraints for the flight trajectory. The vehicle is capable of flying the avoidance 13 trajectory by passing through one or more expected waypoints in specified directions. The two 14 guidance methods are finally tested in four entry missions constrained by various no-fly zones. 15 Results indicate that these methods are effective for no-fly zone avoidance missions in both the nominal and dispersed cases. 16

17

18 Keywords: Entry guidance; geographic constraint; no-fly zone; dynamic heading corridor; waypoint 19

#### 20 1. Introduction

21 The guidance for aerospace vehicles with no-fly zone constraint is an interesting and challenging 22 problem. The no-fly zone is defined as the area where vehicles are not allowed to fly into [1]. This 23 zone originates from obstacles and threats for Unmanned Aerial Vehicles (UAVs), missiles, or other 24 aerospace vehicles [2-4]. In order to generate a trajectory for a UAV to avoid enemy radars, a 25 planning method was developed by Bortoff [5] based on the Voronoi Graph and the virtual force. 26 Kumar and Ghose [6] proposed a sliding circle guidance algorithm for the obstacle avoidance in a 27 planar flight, and extended it to a three-dimensional flight mission. Yang and Zhao [4] developed four 28 geometric shape models for obstacles, including ellipsoid, cuboid, cylinder, and pyramid. For the 29 mission of multiple UAVs, Richards et al. [7] investigated the trajectory planning problem of the fleet 30 coordination under the no-fly zone constraint.

The no-fly zone constraint for entry vehicles was first paid attention to by Jorris [8]. Typical entry vehicles are the crew exploration vehicle (CEV), the Space Shuttle, the X-33 vehicle, and the

<sup>\*</sup>Corresponding author. Tel.: +86-10-82314573-13.

E-mail address: buaa.gnc@gmail.com (Q. Li).

Download English Version:

# https://daneshyari.com/en/article/5472694

Download Persian Version:

https://daneshyari.com/article/5472694

Daneshyari.com