

Contents lists available at ScienceDirect

Applied Ocean Research



journal homepage: www.elsevier.com/locate/apor

Experimental analysis on the risk of vortex ventilation and the free surface ventilation of marine propellers



Anna Maria Kozlowska^{a,b,*}, Sverre Steen^{a,b}

^a Department of Marine Technology, NTNU, 7491 Trondheim, Trondheim, Norway

^b Rolls Royce University Technology Centre "Performance in a Seaway", Trondheim, Norway

ARTICLE INFO

Article history: Received 21 January 2017 Received in revised form 9 June 2017 Accepted 12 July 2017

Keywords: Vortex ventilation Propeller hull vortex cavitation Boundaries Empirical relations Ventilation inception Model tests

ABSTRACT

The paper presents a discussion of the ventilation inception and air drawing prediction of ships propellers, aiming to predict under what conditions ventilation will happen, and the actual physical mechanism of the ventilation.

Three different types of ventilation inception mechanisms are included in our discussion: free surface vortex ventilation, ventilation by sucking down the free surface without forming a vortex as well as ventilation by propeller coming out of the water. Ventilation prediction is based on a series of model tests, where the propeller is tested in different levels of intermittent ventilation. The use of underwater video gives a visual understanding of the ventilation phenomena.

Ventilation by vortex formation has analogies with other phenomena, such as the inlet vortex in pump sumps, ground vortex at the inlet of the aircraft engines and the Propeller Hull Vortex Cavitation (PHVC). The paper includes comparison between Propeller Hull Vortex Cavitation (PHVC) and Propeller Free Surface Vortex Ventilation (PFSVV) as well as comparison between PFSVV and vortex formations of aero engines during high power operation near a solid surface.

Experimental data based on several different model tests shows the boundary between the vortex forming, non-vortex forming and free surface ventilation flow regimes. For comparison the following parameters, which determined the intensity of the hydrodynamic interaction between the propeller and free surface have been used: propeller load coefficient c_T , tip clearance ratio c/D, propeller submergence ratio h/R, ambient velocity V_i and flow cavitation/ventilation number $\sigma_{cav}/\sigma_{vent}$.

© 2017 Elsevier Ltd. All rights reserved.

1. Introduction

When a ship propeller operates under highly loaded condition, unsteady line vortex cavitation may occur between the propeller tip and the hull. This type of cavitation is known as propeller – hull vortex cavitation (PHVC) and, if it occurs, it causes strong vibrations and noise in the stern of the ship. When a propeller is operating close to the free water surface, a vortex might form between the propeller and the free surface through which air can be drawn down to the propeller, so that it ventilates – a phenomenon we call Propeller Free Surface Vortex Ventilation (PFSVV). Ventilation typically occurs when the propeller loading is high and the propeller are

E-mail addresses: anna.kozlowska@ntnu.no (A.M. Kozlowska), sverre.steen@ntnu.no (S. Steen).

http://dx.doi.org/10.1016/j.apor.2017.07.006 0141-1187/© 2017 Elsevier Ltd. All rights reserved. large due to heavy seas. Propeller ventilation inception depends on different parameters i.e. propeller loading, forward speed and the distance from the propeller to the free surface, see for instance Califano [2], Kozlowska et al. [9] and Kozlowska and Steen [10]. It is likely that the physical phenomena causing vortex forming of PHVC and vortex ventilation are closely related, see Huse [5]. In this paper, PFSVV will be compared to PHVC with the aim of getting a better understanding of the physical mechanisms causing PFSVV, and on that basis enable the making of better simulation and prediction methods for PFSVV.

Ventilation by vortex formation (PFSVV) has been studied by several researchers see for instance Koushan (2006 I, II and III), Kozlowska et al. [9], Kozlowska and Steen [10], Califano [2], Koushan et al. [8] and Kozlowska et al. [11].

Kozlowska et al. [9] focused on ventilation inception mechanisms, classification of different types of ventilation, thrust loss related to each type of ventilation, and provided a simple calculation method for predicting thrust loss.

^{*} Corresponding author at: Department of Marine Technology, NTNU, Trondheim 7491, Norway.

2	A.M. Kozlowska, S. Steen / Ap
Nomen	clature
Symbols	sindex
a_v	Vortex radius [m]
с	Tip clearance, distance from the top of propeller disk
	to the surface (hull) [m]
c/D	Tip clearance ratio [-]
c_T	Propeller load coefficient [-]
c _{Tn}	Propeller load coefficient for non-ventilated deeply
	submerged propeller [–]
<i>c</i> _{0.7}	Chord length at 0.7R [m]
<i>c</i> _{<i>L</i>0.7}	Lift coefficient at 0.7R [m]
D, R	Propeller diameter, propeller radius [m]
h	Propeller submergence from the propeller axis to
	the free surface [m]
h/R	Propeller submergence from the propeller axis to
_	the free surface [–]
J	Advance number [–]
Jc	Critical advance coefficient [-]
Jsc	Super critical advance coefficient [-]
K _T	Thrust coefficient [–]
K _{Tn}	Time-averaged mean value of the thrust coefficient
	for deeply submerged non-ventilated propeller [–]
n D/D	Propeller revolutions [Hz]
P/D	Propeller pitch ratio [-]
PFSVV	Propeller free surface vortex ventilation [–]
РНVС Г	Propeller hull vortex cavitation [–]
	Span-wise circulation [m ² /s]
p_{ν}	Vapor pressure [Pa]
p_0	Atmospheric pressure [Pa]
S T	Surface tension of the water [N/m]
	Propeller thrust [N]
V_A V_i	Speed of advance [m/s] Velocity through the propeller disk [m/s]
V_0	Free stream velocity [m/s]
z	Number of blades [–]
β_T	Total thrust loss factor [–]
σ_{cav}	Cavitation number [–]
	Ventilation number [–]
σ_{vent}	Density of water [kg/m ³]
ρ ν	Kinematic viscosity [m ² /s]
-	D Advance coefficient
$K_T = \frac{1}{\rho n}$	
1	$\frac{K_T}{l^2}$ Propeller load coefficient
	(\vec{R}) Tip clearance, distance from the top of propeller
	disk to the free surface (hull)
$\sigma_{cav} = \frac{1}{0}$	
	erence pressure for the cavitation number)

 $\sigma_{vent}=2gh/(V_{\infty})^2$ Ventilation number $We = nD\sqrt{\rho D/S}$ Weber number

Kozlowska and Steen [10] focused on comparison between ventilation in static and dynamic conditions (heave motion) both for open and ducted propeller, and discussed how to estimate thrust loss. As a conclusion, a new formulation of the relations between ventilation and thrust loss was developed.

Kozlowska et al. [11] presented comparison between model tests and numerical calculations of thrust loss due to ventilation. The comparison contains two main aspects: comparison between blade forces and moments during non-ventilating and ventilating phase and comparison of results of flow visualization using high speed video (experiments) with CFD simulation results. The com-

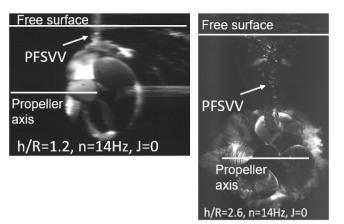


Fig. 1. Impact of the free surface ventilation (PFSVV) for complete submerged propellers.

parisons aim at identifying the degree of correlation and discuss reasons for deviations.

PFSVV occurs for completely submerged, highly loaded propellers at low advance speed. The vortex funnel can reach the surface quite far from the propeller disc, especially for large submergence ratios. Fig. 1 shows two examples of PFSVV. Note that submergence h is the distance from the undisturbed free surface to the propeller axis.

The PHVC phenomenon was first reported by Huse [5]. Systematic observations had been carried out to investigate the effect of the afterbody form, tip clearance c/D, propeller loading c_T and cavitation number σ_{cav} . Experimental observation with a flat, horizontal plate above the propeller in a cavitation tunnel showed that PHVC is more likely to occur for small tip clearances (up to 20% of propeller diameter, c = 0.2D) for low advance coefficient J.

Based on experimental investigations four hypotheses have been suggested for criteria leading to PHVC: a so called "starting vortex", "vortices created by the shear flow in the wake field", "vortices created in other regions of the flow field" as well as "the pirouette effect", see Fig. 2. The "Starting vortex" hypothesis is based on Helmholtz's second theorem, which states that a vortex must be either closed or terminate on the boundary of the fluid. Fig. 2 below shows the corresponding vortex line representation of a propeller blade. Circulation will also be closed on the shortest possible way. This means that the tip clearance must be less than the blade length and axial flow velocity in the region between hull and blade tip should be close to zero. Hypothesis based on "vortices created by shear flow in the wake field" means that a high wake peak in the upper part of the propeller disk gives rise to intense shear flow in the region of highest velocity gradient. This represents a vorticity in the flow field that may "curl up" to form the concentrated vortices necessary to create PHVC.

The basic idea for the hypothesis based on "vortices created in other regions of the flow field" is that the cores of vortices will cavitate when entering the low pressure region between propeller and hull.

Huse [5] concluded that the hypothesis based of the "pirouette effect" is probably the most correct. By this hypothesis the effect of tip clearance, randomness, effect of blade angular position and effect of vertical fins can be satisfactory explained. The basic phenomena related to "pirouette effect" were further explain later by Martio et al. [12]. As the gap between the propeller blade tip and the wall is decreased, the blade suction side does not obtain enough water from the inlet side, so water is also sucked from downstream, causing a rotation of the flow, which is concentrated into a vortex by the so-called pirouette effect (rotational velocity has to increase considerably in order to keep the angular momentum constant,

Download English Version:

https://daneshyari.com/en/article/5473280

Download Persian Version:

https://daneshyari.com/article/5473280

Daneshyari.com