



# A comparison of micro gas turbine operation modes for optimal efficiency based on a nonlinear model



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## ABSTRACT

The novel contribution of this paper is that Micro gas turbine (MGT) operation modes for optimal efficiency are compared based on a nonlinear model, and the variable-speed control is proposed for optimal efficiency. The nonlinear mathematical MGT model is established based on thermodynamic analysis, which can completely reflect the MGT operational characteristics. When the air flow rate is fixed, the rotational speed of the rotor greatly influences the MGT efficiency. At a certain value of speed, the system efficiency reaches its maximum. On this basis, the efficiency of four MGT operation modes are studied: 1. constant speed of a simple cycle, 2. variable speed of a simple cycle, 3. constant speed of a regenerative cycle, and 4. variable speed of a regenerative cycle. In this paper, the relationship between optimal efficiency and the corresponding rotational speed of different output powers formulated using a numerical calculation method is studied. The optimal efficiency formula can be used to generate the given speed of the MGT speed controller for optimally efficient operation. The results show that the variable-speed operation mode of the regenerative cycle exhibited the highest system efficiency and has an evident efficiency optimization effect under a small load.

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## 1. Introduction

Micro gas turbines (MGTs) using simple cycles and multiaxis structures first appeared in the 1940s. Because of their lack of regenerators and heavy gear box, their efficiency is lower than that of internal combustion engines. These disadvantages have limited the development of MGTs. With the development of high-efficiency compact heat exchangers, high-speed pneumatic bearings, high-speed permanent magnet generators, and power electronic technology, the modern advanced MGT has developed rapidly since the 1990s [1]. Compared with the internal combustion engine, the modern advanced MGT has the advantages of a greater ratio of power to volume [2], operation with diverse forms of fuel, lower noise, less pollution, and suitability for combined cooling, heating, and power (CCHP) system [3]. The MGT generation system is an important distributed generation system [4,5], and many scholars have done a lot of research work in turbomachinery performance analysis [6,7], MGT performance analysis [8,9] and so on.

Because of the MGTs high response and high combined cycle

efficiency, the MGT has drawn great attention in both research and application fields. However, in some special applications, the MGT is limited, largely because its performance is not perfect.

Efficiency is an important indicator of performance in MGTs. Improving the operating efficiency of MGTs can reduce costs, which is conducive to increasing their popularity. There are many factors that affect efficiency (e.g., humidity, ambient temperature, fuel type, and cooling technology). Existing research results show that a relative humidity in the range of 30%–45% has less of an impact on efficiency. When the ambient temperature is in the range of 26.7°C–37.8°C, the efficiency increases by 1% for every 5.6°C rise in temperature. Compared with combustion of diesel oil and heavy oil, the efficiency of burning natural gas is the highest [10]. Some researchers have studied the efficiency from the viewpoint of MGT design, demonstrating that the higher the design temperature ratio, the lower the fuel consumption and the higher the part load efficiency. The design value of the pressure ratio has little effect on efficiency [11]. The MGT design is related to the complex internal structure and field operation, which are not the focus of this paper. In this paper, we perform the theoretical research work to improve MGT efficiency from the control perspective.

To study MGT efficiency from the viewpoint of theoretical

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**Nomenclature**

$\alpha$	comprehensive heat transfer coefficient	$c_{pa}$	average specific heat at constant pressure of air
$\alpha_a$	heat transfer coefficient between air and metal wall	$c_{pg}$	average specific heat at constant pressure of gas
$\alpha_g$	heat transfer coefficient between gas and metal wall	$G_{ASC}(s)$	transmission function of anti radiation cover
$\bar{G}_c$	similarity mass flow of compressor	$G_c$	mass flow of compressor
$\bar{G}_t$	similarity mass flow of turbine	$G_f$	mass flow of fuel
$\bar{n}_c$	similarity speed of compressor	$G_t$	mass flow of turbine
$\bar{n}_t$	similarity speed of turbine	$G_{VP}(s)$	transmission function of valve positioner
$\hat{\eta}_c$	reduced efficiency of compressor	$J$	rotor moment of inertia
$\hat{\eta}_t$	reduced efficiency of turbine	$K_{eoni}$	integral gain of speed closed-loop PI controller
$\hat{\pi}_c$	reduced pressure ratio of compressor	$K_{eomp}$	proportional gain of speed closed-loop PI controller
$\dot{G}_c$	reduced mass flow of compressor	$K_F$	feedback coefficient of Flow regulating valve
$\dot{G}_t$	reduced mass flow of turbine	$n$	speed
$\dot{n}_c$	reduced speed of compressor	$p_1$	Inlet total pressure of compressor
$\dot{n}_t$	reduced speed of turbine	$p_2$	outlet total pressure of compressor
$\eta_B$	combustion efficiency	$p_3$	input pressure of turbine
$\eta_c$	efficiency of compressor	$p_4$	outlet total pressure of turbine
$\eta_{MGT}$	efficiency of MGT	$P_a$	atmospheric pressure of environment
$\eta_t$	efficiency of turbine	$P_c$	consumption power of compressor
$\kappa_a$	isentropic exponent of air	$P_f$	power of fuel
$\kappa_g$	isentropic exponent of gas	$P_{ML}$	load power
$\pi_c$	pressure ratio of compressor	$P_m$	output power of MGT
$\pi_t$	expansion ratio of turbine	$P_t$	output power of turbine
$\sigma$	regenerator effectiveness	$Q_u$	low heat value of fuel
$\tau_F$	time constant of Flow regulating valve	$t$	time
$\tau_T$	integral ratio of temperature controller	$T_1$	inlet temperature of compressor
$A$	comprehensive heat transfer area	$t_1$	relative coefficient of turbine
$a$	coefficient of valve positioner	$T_2$	output temperature of compressor
$a_1$	inlet total pressure recovery coefficient of compressor	$T_2'$	temperature of cold outlet end
$A_a$	heat transfer area between air and metal wall	$T_3$	input temperature of turbine
$A_g$	heat transfer area between gas and metal wall	$T_4$	output temperature of turbine
$b$	coefficient of valve positioner	$T_4'$	temperature of hot outlet end
$c$	coefficient of valve positioner	$T_m$	metal wall average temperature of recuperator
$c_1$	relative coefficient of compressor	$T_R$	temperature setpoint
$c_2$	relative coefficient of compressor	$Z$	constant
$c_3$	relative coefficient of compressor	*	refers to the stagnation parameter
		ref	reference value of the corresponding variable

derivation and numerical simulation, it is necessary to set up a mathematical model of the MGT. Widely used in the existing literature are the Rowen model [12–15], the IEEE model [16–19] and component-based models that are widely used for performance simulation of turbomachinery [20]. In these two models, each link of the gas turbine is simplified as a linear or a delay link in the near-design condition, and gas flow, compressor pressure ratio, turbine expansion ratio, efficiency, and other process variables are not included. In addition, the models do not include the regenerator; therefore, the accuracy for off-design operation is poor. Other models, such as the aero-derivative model [21], GAST model [22], WECC model [23–25], CIGRE model [26], frequency-dependent model [27], and fluid network model [28–30], are too simple to accurately describe the efficiency and output power characteristics of the MGT under the off-design condition. Furthermore, the measured curve is difficult to describe mathematically. The efficiency of the compressor and turbine under the off-design condition has been presented in the published literature and can be expressed as an analytic formula relating speed and gas flow [31–34]. At the same time, a mathematical model of the regenerator has been considered [35], so a nonlinear mathematical model of the MGT has been established.

Compared with the existing nonlinear mathematical model, the

model used in this paper has the following characteristics: 1) It uses a static average model of the heat exchanger, and 2) the analytical formula for the efficiency of the compressor and turbine under the off-design condition is adopted [36]. The nonlinear mathematical model of the MGT used in this paper is that proposed by the authors of [36]. On this basis, under different load conditions the relationship between speed and efficiency is studied. The curve of efficiency and control variables can be drawn using these mathematical relations. The operating point of the optimal efficiency can be clearly demonstrated from the variation of the curve.

The major work of this paper is to analyze the difference in the optimal efficiency operation law for two different MGT structures of simple and regenerative cycles. There are two operation modes for each structure common operation and optimal efficiency operation so there are four different operations to study. In this paper, the efficiency of the four operation modes is calculated quantitatively to identify the best operation mode in the whole operational condition range, and the degree of efficiency improvement is given. We obtain the relationship between the load and the variable speed for optimal efficiency, which is applied in the control system.

The article is organized as follows: Section 2 presents the nonlinear mathematical MGT model. In Section 3, we analyze the optimal efficiency operation. Sections 4 and 5 present the optimal

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