



Research paper

Preliminary design criteria of Stirling engines taking into account real gas effects



Fernando Sala ^a, Costante Invernizzi ^a, David Garcia ^b, Miguel-Angel Gonzalez ^b,
Jesús-Ignacio Prieto ^{b,*}

^a University of Brescia, Via Branze 38, 25123 Brescia, Italy

^b University of Oviedo, Campus de Viesques, 33204 Gijón, Spain

H I G H L I G H T S

- The theoretical limit of the gas circuit performance considers real gas effects.
- The influences of physical gas properties are analysed.
- Simulations predict the feasibility of operating under real gas effects.
- Equations and correlations are proposed as preliminary design criteria.
- The procedure is supported by an experimental database.

A R T I C L E I N F O

Article history:

Received 15 December 2014

Accepted 26 June 2015

Available online 6 July 2015

Keywords:

Stirling engine

Preliminary design

Real gas effect

Dimensional analysis

Performance correlations

A B S T R A C T

This article deals with the main geometric parameters, operating variables and experimental results of engines with different size and characteristics, as well as simulation predictions at operating conditions for which the working fluid may evidence real gas effects. The concept of dimensionless quasi-static indicated work, introduced previously, is computed assuming both ideal and real gas models. Both values are compared with the Schmidt's model prediction, to evaluate separately how the mechanism simplification and the equation of state affect. The influence of physical gas properties are also analysed regarding the gas circuit performance and mechanical efficiency. Pending of experimental corroboration, some simulations show the possibility of obtaining interesting operating conditions under real gas effects.

Semi-empirical equations and experimental correlations are proposed as preliminary design criteria. The ratio between the dimensionless values of the maximum indicated power (a sort of 'indicated' Beale number) and the quasi-static indicated work is analysed. The procedure to estimate the maximum indicated power is completed by analysing the characteristic Mach number (dimensionless rotation frequency). Analogous procedure is applied for the maximum brake power and corresponding engine speed. Use of dimensionless variables facilitates the generalization of analyses by means of dynamic similarity criteria.

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1. Introduction

Potential benefits justify that the Stirling engine has competed with other technologies over the past three decades for applications such as cogeneration at industrial and domestic scales, underwater systems and air independent propulsion of submarines,

solar thermal energy conversion, and space vehicles. The commercial battle is full of ups and downs, since the cost of the product is mainly determined by the volume of series production of its components, and by fluctuations in the price of fossil fuels. So far, the Stirling cycle has achieved greater commercial success in their applications as reverse cycle machine, an option that other thermal engines do not have.

The gas circuit and the drive mechanism are the main sub-systems of a Stirling engine. The numerical computing power currently available allows the development of models to study

* Corresponding author.

E-mail address: jprieto@uniovi.es (J.-I. Prieto).

Nomenclature

A_{cx}	cross-sectional area of space x (m^2)	T_x	working gas temperature at space x (K)
C_f	local, instantaneous friction factor $= \Delta p \cdot 2\rho(A_{xx}/\dot{m})^2(r_{hx}/L_x)$	V	volume (m^3)
c_p	constant pressure heat capacity (J/kg K)	V_C	swept volume of the compression piston (m^3)
c_v	constant volume heat capacity (J/kg K)	V_{dx}	dead volume of space x (m^3)
h	convective heat transfer coefficient (W/m^2 K)	V_E	swept volume of the expansion piston (m^3)
k	thermal conductivity (W/m K)	V_{sw}	net swept volume = $V_{max} - V_{min}$ (m^3)
L_x	length of space x (m)	W_0	quasi-static indicated work per cycle (J)
\dot{m}	mass flow rate (kg/s)	$\alpha, \beta, \chi, \delta$	dimensionless coefficients of mechanical power losses
m_1	reference mass of drive mechanism moving parts (kg)	α	phase angle (rad)
N_B	Beale number	α_R	regenerator material thermal diffusivity (m^2/s)
N_m	characteristic drive mechanism number = $m_1 R T_C / (p_m V_{sw})$	α_{cx}	dimensionless cross-sectional area of space $x = A_{xx}/V_{sw}^{2/3}$
N_{MA}	characteristic Mach number = $n_s V_{sw}^{1/3} / \sqrt{RT_C}$	$\delta_1 \dots \delta_n$	dimensionless geometrical parameters, including those characteristic of the drive mechanism
N_{ma}	local, instantaneous Mach number = $\dot{m} \sqrt{RT_C} / (p A_{xx})$	η_{mec}	mechanical efficiency = P_B/P_{ind}
$N_{MA,max}$	Mach number corresponding to the maximum indicated power = $n_{s,max} V_{sw}^{1/3} / \sqrt{RT_C}$	$\eta_{mec,max}$	mechanical efficiency at $n_{s,max}^*$
$N_{MA,max}^*$	Mach number corresponding to the maximum brake power = $n_{s,max}^* V_{sw}^{1/3} / \sqrt{RT_C}$	Φ	dimensionless factor of linear losses of indicated power
$N_{ma,max}$	local Mach number corresponding to the maximum mass flow rate	γ	adiabatic coefficient = c_p/c_v
N_p	characteristic pressure number = $p_m V_{sw}^{1/3} / (\mu \sqrt{RT_C})$	κ	swept volume ratio = V_C/V_E
N_{PR}	characteristic Prandtl number = $\mu c_p/k$	κ_s	swept volume ratio according to Iwamoto et al. = V_{sw}/V_E
N_{pr}	local, instantaneous Prandtl number	λ_{hx}	dimensionless hydraulic radius of space $x = r_{hx}/V_{sw}^{1/3}$
N_{re}	local, instantaneous Reynolds number = $4\dot{m}r_{hx}/(\mu A_{xx})$	μ	working fluid viscosity (Pa s)
$N_{re,max}$	local Reynolds number corresponding to the maximum mass flow rate	μ_{dx}	dead volume ratio of space $x = V_{dx}/V_{sw}$
N_{SG}	characteristic Stirling number = $p_m/(\mu n_s)$	μ_L	lubricant viscosity (Pa s)
n_s	engine speed (rev/s)	Π_V	Regenerator volumetric porosity
$n_{s,max}$	engine speed at maximum indicated power (rev/s)	θ	crank angle (rad)
$n_{s,max}^*$	engine speed at maximum brake power (rev/s)	Ψ	dimensionless factor of quadratic losses of indicated power
N_{st}	local, instantaneous Stanton number = $h A_{xx}/(\dot{m} c_p)$	ρ	density (kg/m^3)
N_{TCR}	regenerator thermal capacity ratio = $\rho_R c_R / (\rho c_p) = \rho_R c_R T_C / (p_m (\gamma - 1) \gamma)$	ρ_{RCR}	regenerator material volumetric specific heat capacity (J/m^3 K)
N_W	West number	τ	temperature ratio = T_C/T_E
N_α	characteristic regenerator thermal diffusivity number $= \alpha_R / (V_{sw}^{1/3} \sqrt{RT_C})$	ζ_B	dimensionless brake power = $P_B/(p_m V_{sw} n_s)$
p	pressure (Pa)	ζ_{ind}	dimensionless indicated power = $P_{ind}/(p_m V_{sw} n_s)$
P_B	brake power (W)	ζ_{mec}	dimensionless mechanical power losses = $P_{mec}/(p_m V_{sw} n_s)$
p_{cr}	pressure at the critical point (Pa)	ζ_0	dimensionless quasi-static work per cycle = $W_0/(p_m V_{sw})$
P_{ind}	indicated power (W)		
p_m	mean pressure (Pa)		
P_{mec}	mechanical power losses = $P_{ind} - P_B$ (W)		
p_r	reduced pressure = p/p_{cr}		
p_{rm}	reduced mean pressure = p_m/p_{cr}		
R	specific gas constant (J/kg K)		
r_{hx}	hydraulic radius of space x (m)		
T_{cr}	temperature at the critical point (K)		
T_{rx}	reduced temperature at space $x = T_x/T_{cr}$		

Subscripts

C	cooler or compression space
cc	compression cylinder
E	heater or expansion space
ec	expansion cylinder
R	regenerator
x	generic space

Superscripts

R	considering real gas effects
S	according to Schmidt cycle

these subsystems with fewer simplifying assumptions than was possible decades ago. Thus, the development of Stirling engine models capable of being integrated into schemes for the overall techno-economic analysis of applications is still a topic of current interest [1–6]. Because all models must be validated experimentally, the benefit of the advanced models is not greater accuracy but the ability to analyse physical phenomena dependent on variables whose experimental measurement is practically impossible. For

now, simple models are needed at the preliminary design stage, while the advanced models can be appropriate for optimisation tasks.

Beale and West numbers were probably the simplest and most commonly used criteria for sizing of engines at the preliminary design stage. However, without prejudice to its historical importance, it should be recognized that both concepts are experimental correlations where great simplifications have been made. So far it

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