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IFAC-PapersOnLine 49-7 (2016) 347–352

### Active Compressor Surge Control System hy Using Piston Actuation: Implementation  $_3$  1 iston Actuation. Implem<br>and Experimental Results  $^\star$  $\frac{1}{2}$  Active Compressor Surge Control System Syst Active Compressor Surge Control System Active Compressor Surge Control System Active Compressor Surge Control System g Piston Actuation: Implem by Using Piston Actuation: Implementation by Using Piston Actuation: Implementation

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**Abstract:** A nover implementation and experimental test results of a piston actuated active<br>surge control system (PAASCS) on a laboratory scale pipeline-compressor system are presented. The experimental test is done to prove the concept of stabilizing compressor surge by dissipating the plenum energy using a piston actuation. The PAASCS's controller is applying dissiparing the pienuin energy using a piston actuation. The TAASCS S controller is applying  $\psi$ -control introduced in (Uddin and Gravdahl, 2016), which only uses feedback from pressure p-control introduced in (Gudin and Gravdam, 2010), which only uses reedback from pressure<br>measurements at the compressor discharge and in the plenum. Practical aspects of implementing  $\psi$ -control introduced in (Uddin and Gravdahl, 2016), which only uses feedback from pressure measurements at the compressor discharge and in the plenum. Practical aspects of implementing the PAASCS are presented includin on a compressor performance test, piston design, and the test setup. The experimental test on a compressor performance test, piston design, and the test setup. The experimental test<br>results show that the PAASCS is able to stabilize surge and prove the concept of PAASCS with results show that the FAASCS is able to stabilize surge and prove the concept of FAASCS with<br>the advantage of  $\psi$ -control which stabilizes compressor surge by using feedback from pressure measurements only. Abstract: A novel implementation and experimental test results of a piston actuated active surge control system (PAASCS) on a laboratory scale pipeline-compressor system are presented. the advantage of  $\psi$ -control which stabilizes compressor surge by using feedback from pressure measurements only.

© 2016, IFAC (International Federation of Automatic Control) Hosting by Elsevier Ltd. All rights reserved.  $\sigma$  sort, and  $\mu$  and  $\sigma$  is the complete surface flow measurement, linear actuator, linear actuator, linear actuator,  $\sigma$  $\odot$  2016 IFAC (Intern

*Keywords:* Compressor, active surge control, pitot tube flow measurement, linear actuator, hardware in the loop test, experimental. hardware in the loop test, experimental. hardware in the loop test, experimental. hardware in the loop test, experimental.

# 1. INTRODUCTION 1. INTRODUCTION 1. INTRODUCTION 1. INTRODUCTION

A centrifugal compressor operating area is commonly A centrifugar compressor operating area is commonly<br>shown by a compressor map, where the compressor operation at low mass flows is limited by a surge line. The eration at low mass hows is filled by a surge line. The<br>operating area on the left side of the surge line is unstable operating area on the left side of the surge line is unstable<br>and will lead to surge, while it is stable operating area on the right side of the line. Compressor surge is an on the right side of the line. Compressor surge is an and will lead to surge, while it is stable operating area<br>on the right side of the line. Compressor surge is an<br>aerodynamic instability in the compression system and aerodynamic instability in the compression system and<br>results in an axisymmetric oscillation of the compressor results in an axisymmetric oscillation of the compressor<br>mass flow and the compressor pressure. The instability is mass now and the compressor pressure. The instability is<br>physically indicated by pressure fluctuation, reversal flow, physically indicated by pressure fluctuation, reversal flow,<br>temperature fluctuation and followed by severe vibration. Compressor surge leads to compressor damage especially compressor surge reads to compressor damage especially<br>at the rotating parts, for examples: compressor blades, at the rotating parts, for examples. compressor blades,<br>shaft and bearing, and also pipeline and structure (Gravdahl and Egeland, 1999). on the right side of the line. Compressor surge is an dahl and Egeland, 1999). dahl and Egeland, 1999). dahl and Egeland, 1999).

A method to stabilize surge by using a state feedback con-A method to stabilize surge by using a state reedback con-<br>trol was introduced by Epstein et al. (1986). The method is known as active surge control system (ASCS). Several is known as active suge control system (ASCS). Several<br>actuators have been applied in the ASCS as summarized in (Willems and de Jager, 1999; Uddin and Gravdahl, 2015), (Willems and de Jager, 1999; Uddin and Gravdahl, 2015), (Willems and de Jager, 1999; Uddin and Gravdahl, 2015), (Willems and de Jager, 1999; Uddin and Gravdahl, 2015), (Willems and de Jager, 1999; Uddin and Gravdahl, 2015),<br>for examples: movable plenum wall, close couple valve, for examples. Inovable plenum wan, close couple varve, drive torque, active magnetic bearing, and piston actuation. Based on how the actuators work to stabilize surge, ation. Based on how the actuators work to stabilize surge,<br>the ASCS can be classified in two types: upstream energy injection and downstream energy dissipation (Uddin and Gravdahl, 2015). The upstream energy injection ASCS drive torque, active magnetic bearing, and piston actu-Gravdahl, 2015). The upstream energy injection ASCS Gravdahl, 2015). The upstream energy injection ASCS Gravdahl, 2015). The upstream energy injection ASCS is stabilizing surge by increasing the upstream pressure to increase the upstream energy, while the downstream to increase the upstream energy, while the downstream<br>energy dissipation ASCS is stabilizing surge by flowing energy dissipation ASCS is stabilizing surge by nowing extra nuit out nom the pienum to decrease the down-<br>stream energy. Two general state feedback control law for the both ASCS types were introduced by Uddin and Gravdahl (2016), and named  $\phi$ -control for the upstream  $\alpha$  existentially expression and  $\psi$ -control for the down stream energy energy injection and  $\psi$ -control for the down stream energy dissipation. Both control laws make the closed loop system of them are the minimum sensor requirement. The  $\phi$ of them are the minimum sensor requirement. The  $\phi$ -<br>control requires feedback from the compressor mass flow  $\frac{1}{2}$  control requires feedback from the compressor mass now measurement only, while the ψ-control requires feedback from pressure measurements at the compressor discharge and in the plenum only. and in the plenum only. from pressure measurements at the compressor discharge from pressure measurements at the compressor discharge and in the plenum only. stream energy. Two general state feedback control law<br>for the hoth ASCS types were introduced by Uddin and measurement only, while the  $\psi$ -control requires feedback<br>from pressure measurements of the somewords discharge and in the plenum only. and in the plenum only.

Piston actuated active surge control system (PAASCS) Piston actuated active surge control system (PAASCS)<br>was in introduced by Uddin and Gravdahl (2011b). It is was in infroduced by Oddin and Gravdam (2011b). It is<br>in the class of ASCS with downstream energy dissipation. Theoretical works to improve the PAASCS performance Theoretical works to improve the PAASCS performance<br>have been done by: introducing integral action to eliminate nave been done by. introducing integral action to eminiate<br>piston drift (Uddin and Gravdahl, 2011a), and introducing piston drift (Uddin and Gravdahl, 2011a), and introducing<br>a back-up system by using blow-off mechanisms (Uddin a back-up system by using blow-on mechanisms (Oddinated System and Gravdahl, 2012b) and by using surge avoidance system (SAS) (Uddin and Gravdahl, 2012a) for fail-safe operation. (SAS) (Uddin and Gravdahl, 2012a) for fail-safe operation. (SAS) (Uddin and Gravdahl, 2012a) for fail-safe operation. (SAS) (Uddin and Gravdahl, 2012a) for fail-safe operation.

This paper presents the implementation and the experimental test results of a PAASCS on a laboratory scale pipeline-compressor system running at a constant compressor speed. An experimental test setup is built in pressor speed. All experimental test setup is built in<br>Compressor laboratory at Departement of Engineering Cybernetics, NTNU. The PAASCS control law is applying This paper presents the implementation and the exper-<br> $\frac{1}{2}$  test results of a PAACCS on a laboratory scale

 $\star$  The authors acknowledge the financial support of Siemens Oil and Gas Solutions Offshore through the Siemens-NTNU research collaboration project. and Gas Solutions Offshore through the Siemens-NTNU research collaboration project. collaboration project. collaboration project. and Gas Solutions Offshore through the Siemens-NTNU research

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Fig. 1. Piston actuated active surge control system.

to build a hardware in loop simulation. The PAASCS is tested experimentally to see the performance of the system formance test. The control law algorithm is written in formance test. The control law algorithm is written in *the test setup parameters* and a compressor map. The compressor map is obtained through a compressor per-*Lc* in stabilizing surge. The test is started by running the In a wagorithm is written in<br>Matlab and Simulink and embedded in a dSpace board the  $\psi$ -control. The control gain is determined by using the test setup parameters and a compressor map. The compressor at a constant speed and reducing the flow by closing an outlet valve (throttle) such that the compressor is entering surge while the PAASCS is inactive. The PAASCS is then activated after the compressor is in surge. The surge is shown by pressures oscillation sensed by pressure sensors at the compressor discharge and in the plenum. This is the first experimental confirmation of actively stabilizing compressor surge by using piston actuation.

# 2. SYSTEM DYNAMICS AND CONTROL

A model of PAASCS is shown in Figure 1. The PAASCS model is a modification of the Greitzer compressor model (Greitzer, 1976) by adding a piston. The assumptions in the Greitzer model are applied in the PAASCS model. The  $p_A$  and  $p_B$  are the ambient pressures and assumed to be equal, and fluid pressures in the system are measured relative to the ambient pressure. It is assumed that pressure drop along the ducts in the system are neglected. Therefore, the compressor discharge pressure  $(p_d)$  is equal to the compressor pressure rise  $(p_c)$ , and the compressor discharge mass flow  $(w_d)$  is equal to the inlet mass flow  $(w_i)$ . Dynamic equations of the PAASCS model are given as follows (Uddin and Gravdahl, 2011b):

$$
\dot{w}_i = \frac{A_c}{L_c}(p_c - p_p) \tag{1}
$$

$$
\dot{p}_p = \frac{a_0^2}{V_p}(w_i - w_o - w_s),\tag{2}
$$

where  $A_c$  is the compressor duct cross-sectional area,  $L_c$  is the effective length of the equivalent compressor duct,  $p_p$ is the plenum pressure,  $a_0$  is the speed of sound,  $V_p$  is the plenum volume,  $w<sub>o</sub>$  is the outlet mass flow, and  $w<sub>s</sub>$  is the piston mass flow. The outlet mass flow is the set point of the desired compressor operating mass flow. A compressor operating point is an equilibrium point where  $\dot{w}_i = 0$  and  $\dot{p}_p = 0$ . The piston mass flow is defined by:

$$
w_s = \rho A_s \frac{dL_s}{dt},\tag{3}
$$

Table 1. Major components of test setup.

Component	Detail
Compressor	Supercharger Vortech V-1 S-Trim Race M
Pressure sensor	Druck PTX 610
Mass flow sensor	Endress+Hauser t-mass 65F80
Valve	Siemens PN 10
Pipeline	Polypropylene pipe with diameter 75mm
Plenum	Cylindrical vessel
Control board	dSpace DS1103
Piston	See Section 3.3

where  $\rho$  is the fluid density,  $A_s$  is the piston cross-sectional area, and  $L_s$  is the piston position. To simplify the surge control design, we ignore the piston dynamic and assume that the piston will generate a mass flow  $w_s$  following a reference signal. Define constants  $B_1 = \frac{A_c}{L_c}$  and  $B_2 = \frac{a_0^2}{V_p}$ , and substitute them into  $(1)$  and  $(2)$  such that results in:

$$
\dot{w}_i = B_1(p_c - p_p) \tag{4}
$$

$$
\dot{p}_p = B_2(w_i - w_o - w_s). \tag{5}
$$

The PAASCS is in the class of downstream energy dissipation ASCS (Uddin and Gravdahl, 2011b) such that the  $\psi$ -control introduced in (Uddin and Gravdahl, 2016) is applicable.

*Theorem 1.* The  $\psi$ -control states that an equilibrium point of (4) and (5) is globally asymptotically stable (GAS) if  $w_s = w_u$ , where  $w_u = -k_u(p_c - p_p)$  with  $\frac{2B_1}{B_2}k_m < k_u$  $\frac{2B_1}{B_2}k_n, k_m = \frac{\partial p_c}{\partial w_i}$  $\Big|_{\text{max}}$  and  $k_n = \frac{\partial p_p}{\partial w_o}$  $\Big|_{\min}$ .

The stability proof can be found in (Uddin and Gravdahl, 2016). The surge control requires  $w_s$  to behave as  $w_u$ , and it is the task of a piston as the actuator of PAASCS.

## 3. LABORATORY TEST SETUP

The concept of PAASCS is implemented and tested on a laboratory scale pipeline-compressor system at Compressor Laboratory in Department of Engineering Cybernetics, NTNU. A PAASCS test setup is built as shown in Figure 2. Major components in the setup are listed in Table 1 and most of the components have been used in an experimental work on compressor surge control using torque drive (Bøhagen, 2007). Fluid pressures in the setup are measured relative to the ambient pressure.

#### 3.1 Mass flow measurement

The mass flow sensor used in the setup is Endress+Hauser t-mass 65F80-AE2AG1AAAAAA. It is a high performance mass flow sensor for industrial gases and compressed air. The sensor is configured to measure mass flow at a range of 0 to 0.26 kg/s. The sensor has high accuracy with measurement error 2% of the measured value (Endress+Hauser, 2015). However, the sensor response is quite slow such that it is only applicable for measuring steady flow and not for measuring unsteady flow. Because surge is unsteady flow, we measure mass flow using a pitot tube as an alternative solution. The pitot tube is installed at the compressor inlet duct and equipped with two pressure sensors to measure the total pressure  $(p_t)$  and the static pressure  $(p_s)$ . The mass flow is calculated by following equation:

$$
\bar{w} = A_c \sqrt{2\rho (p_t - p_s)},\tag{6}
$$

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