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# Surface integrity and fatigue behavior when turning $\gamma$ -TiAl alloy with optimized PVD-coated carbide inserts

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### **KEYWORDS**

- 14 Fatigue behavior; Microhardness; 15 16 Residual stress: 17 Surface roughness; γ-TiAl alloy; 18 19 Tool wear
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#### 1. Introduction 21

Light and thermal-stability alloy materials have received atten-22 23 tions in the aeroengine field.  $\gamma$ -TiAl alloy is mainly formed by

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titanium and aluminum. Ti and Al elements in y-TiAl alloy can effectively improve its mechanical and thermodynamic performance. y-TiAl alloy is used in engine components which are working under 800 °C temperature, instead of nickel-based superalloy.  $\gamma$ -TiAl alloy has good mechanical properties, for example, high specific strength, elastic modulus, low density, and good oxidation resistance, and these characteristics make it suitable in the aircraft engine area, including turbines, compressor blades, etc.<sup>1,2</sup> Its low density can effectively reduce the weight of an engine, thus reducing the inertia of rotating parts, which is beneficial to the rotating parts. Despite that  $\gamma$ -TiAl alloy has attractive mechanical and thermal properties, it also

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Abstract This paper presents a comprehensive investigation on the effects of tool and turning parameters on surface integrity and fatigue behavior in turning  $\gamma$ -TiAl alloy. The wear of inserts

surface, cutting forces, and surface roughness were studied to optimize PVD-coated carbide inserts.

Surface topography, residual stresses, microhardness, and microstructure were analyzed to charac-

terize the surfaces layer under different turning parameters. Surface integrity and fatigue life tests of

 $\gamma$ -TiAl alloy were conducted under turning and turning-polishing processes. The results show that

compared to CNMG120412-MF4, CNMG120408-SM is more suitable because it obtained low cutting force, surface roughness, and tool wear. With increasing the cutting speed and depth, the depths of the compressive residual stress layer, hardening layer, and plastic deformation layer increased. For turning and turning-polishing specimens, the compressive residual stress was relaxed by less than 20-30% after  $10^7$  cycles. The fatigue life of a turning-polishing specimen with  $R_a = 0.15 \,\mu\text{m}$  has increased 3 times from that of a turning specimen with  $R_a = 0.43 \,\mu\text{m}$ .

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has some bad material properties, such as poor ductility, thermal conductivity, fracture toughness, high brittleness, and easiness to react with tool materials.<sup>3</sup> The main problems of y-TiAl are lamellae deformation, surface micro cracks, and higher surface roughness and residual stresses.<sup>4</sup> These defects will become initial crack extension points, eventually leading to a workpiece failure.

Domestic and overseas scholars have carried out a lot of 43 research on the surface integrity of turning  $\gamma$ -TiAl alloy. 44 Zhang<sup>5</sup> have found that the machining ability of an uncoated 45 negative rake hard alloy cutter is better. Adhesion wear is the 46 main wear form of cutting tools. Mantle and Aspinwall<sup>6,7</sup> have 47 48 found that when turning  $\gamma$ -TiAl alloy, the defects of surface 49 integrity include material pull out, surface drag, and surface hardening and cracking. A crack will reduce fatigue life, and 50 there is no expansion of the crack; optimizing processing 51 parameters can reduce surface defects. Sharman et al.<sup>8</sup> have 52 53 found that surface damages included lamellae deformation, 54 surface hardening, and broken TiB<sub>2</sub>. Under the conditions of little cutting depth and feed, the cracks size could be reduced. 55 The lubrication condition and cutting speed had little influence 56 on cutting forces, while the cutting depth had the most influ-57 ence on all three cutting forces. Klocke et al.9,10 have system-58 atically studied the influences of different turning parameters, 59 insert geometry, and lubrication conditions on the cutting sur-60 face roughness and tool wear. Considering the surface quality, 61 62 machining of  $\gamma$ -TiAl can be intensified by changing the parameters and insert geometry. The cutting depth has a smaller 63 effect on the surface roughness than the corner radius  $r_{\varepsilon}$ . When 64 the feed is reduced, the roughness will decrease with an 65 approximately linear trend. A higher cutting speed results in 66 more severe tool wear, and higher surface quality has been 67 found when using tools RCMX ( $r\varepsilon = 0.6$  mm) than that using 68 tools CNMA ( $r\varepsilon = 0.8$  mm) in dry conditions. Tools RCMX 69 70 are round, the radial relief angle is 7°, the tolerance grade is 71 M, and X represents the shape of the screw hole. As for tools CNMA, the rhombus apex angle is 80°, the clearance angle is 72 0°, and the form of chip breaker is MA. Beranoagirre et al.<sup>11</sup> 73 have found that the cutting speed affects more on tool wear 74 75 than on feeding and an increase of the cutting speed leads to worse tool wear. Ma<sup>12</sup> has found that the machining ability 76 of physical vapor deposition (PVD)-coated hard alloy cutters 77 is better. Wan et al.<sup>13</sup> reviewed the high-cycle fatigue behavior 78 of  $\gamma$ -TiAl alloy. The formation and microstructure of  $\gamma$ -TiAl 79 alloy are the key factors that affect its fatigue strength. Xue 80 et al.<sup>14</sup> carried out a bending fatigue test on  $\gamma$ -TiAl alloy at 81 room temperature. The results show that when the fatigue 82 cycles are more than  $10^7$ , the fatigue fracture is still occurring, 83 and the surface roughness has little effect on the fatigue life. 84 Fatigue damage often occurs at the stress concentration of 85 the building surface. Lin et al.<sup>15</sup> studied the residual stress 86 relaxation of the surface after shot peening in a fatigue test 87 88 and found that the residual stress produced at the beginning 89 of the test was faster and the residual stress was stable after 90  $10^3$  cycles of fatigue.

In most of these studies, in terms of  $\gamma$ -TiAl, only the rela-91 tionship between surface damage, roughness, and turning 92 parameters are analyzed, lacking studies on the influences of 93 parameters on roughness, residual stress, microhardness, and 94 microstructure. Previous research has not been able to explore 95 the effects of turning parameters on the metamorphic layer, 96 let alone conclusions between turning parameters and surface 97

integrity. The influences of tool and turning parameters on sur-98 face integrity, including residual stress, surface roughness, 99 microhardness, and surface topography, were studied in this 100 paper when turning y-TiAl by using PVD-coated carbide 101 inserts. This study plays a guiding significance on obtaining 102 high surface quality and promotes the applications of  $\gamma$ -TiAl. 103

### 2. Materials and methods

### 2.1. Workpiece material

Extrusion forming  $\gamma$ -TiAl alloy was used in this study. The 106 main chemical components are shown in Table 1.  $\gamma$ -TiAl allow 107 has good mechanical properties under high temperatures,<sup>7</sup> as shown in Table 2. 109

#### 2.2. Experimental details

Cutting force and tool wear tests were conducted to investigate 111 the effects of tool geometric parameters on the cutting force 112 and tool wear under the condition of  $v_c = 30 \text{ m/min}$ , 113  $a_{\rm p} = 0.6$  mm, and f = 0.08 mm/r. Experiments were per-114 formed using a CK7525 numerical controlled lathe. SAND-115 VIK and SECO carbide blades were used in the study. 116 Geometric parameters of SANDVIK and SECO carbide 117 blades are shown in Table 3. The SANDVIK carbide blade 118 is PVD-TiCN coating, the rake angle is  $-13^{\circ}$ , the clearance 119 angle is 0°, the corner radius is 0.8 mm, and the chip breaker 120 is SM style. The SECO carbide blade is PVD-TiAlN coating, 121 the rake angle is  $-13^{\circ}$ , the clearance angle is  $0^{\circ}$ , the corner 122 radius is 1.2 mm, and the chip breaker is MF4 style. 123

The effects of turning parameters on surface roughness and 124 morphology were also obtained when using three different 125 tools. The specific turning parameters are given in Table 4. 126 The influences of turning parameters on the altered layer was 127 studied when using a CNMG120408-SM tool under different 128 turning parameters. A fatigue specimen with a length of 129 52 mm is shown in Fig. 1. As shown in Fig. 1,  $\sigma_{rx}$  describes 130 the residual stress along the x-axis and  $\sigma_{rv}$  is the residual stress 131 along the y-axis. The two kinds of processes are turning and 132 turning-polishing. The turning parameters are rotation speed 133 n = 1000 r/min, feed f = 0.06 mm/r, and axial cutting depth 134  $a_{\rm p} = 0.2 \, {\rm mm}.$ Specimens were machined using 135 CNMG120408-SM cutting tools. The polishing method was 136 used until the turning mark and blasting crater were out of 137 sight. Fatigue tests were conducted on a rotating bending fati-138 gue testing machine OBWP-10000, at room temperature, with 139 a sinusoidal load, the frequency of which was 100 Hz; the cycle 140 stress ratio R = -1; the stress level was 1000 MPa. Stress 141 relaxation tests were conducted on the same testing machine 142 with a stress level of 410 MPa, and residual stresses tests were 143 conducted under different cycle tests. 144

#### 2.3. Cutting force and surface integrity testing 145

Cutting force tests were conducted by a three-component 146 piezo-electric dynamometer type 9255B with a resonant fre-147 quency of 1.7 kHz. The dynamometer was mounted on the 148 worktable under the tool and connected to a charge amplifier 149 Kistler5080A. The amplified cutting forces signal was acquired 150 Download English Version:

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