Chinese Journal of Aeronautics, (2017), xxx(xx): xxx-xxx



CJA 918

17 October 2017

Chinese Society of Aeronautics and Astronautics & Beihang University

Chinese Journal of Aeronautics

cja@buaa.edu.cn www.sciencedirect.com JOURNAL OF AERONAUTICS

Effect of particle size and oxygen content on ignition and combustion of aluminum particles

Yu'nan ZHOU, Jianzhong LIU^{*}, Daolun LIANG, Wei SHI, Weijuan YANG, Junhu ZHOU

State Key Laboratory of Clean Energy Utilization, Zhejiang University, Hangzhou 310027, China

8 Received 8 November 2016; revised 22 February 2017; accepted 18 April 2017

KEYWORDS

- 13 Aluminum;
- 14

3

5

7

g

11

- Flame;
- 15 Ignition and combustion;16 Laser;
- 16 17

18

Thermal analysis

Abstract Particle size and oxygen content are two of the key factors that affect the ignition and combustion properties of aluminum particles. In this study, a laser ignition experimental system and flame test system were built to analyze the ignition and combustion characteristics and the flame morphology of aluminum particles. A thermobalance system was used to analyze the thermal oxidation characteristics. In addition, the microstructure of aluminum was analyzed by scanning electron microscopy. It was found that the oxidized products were some of the gas phase products agglomerated. Smaller particle size samples showed better combustion characteristics. The combustion intensity, self-sustaining combustion time and the burn-off rate showed a rising trend with the decrease in the particle size. Increasing the oxygen content in the atmosphere could improve the ignition and combustion. Small particle size samples had a larger flame height and luminance, and the self-sustaining combustion time was much longer. Three distinct stages were observed during the thermal oxidation process. The degree of oxidation for small-sized samples was significantly higher than that for the larger particle size samples. Moreover, it was observed that the higher the oxygen content, the higher the degree of oxidation was.

© 2017 Production and hosting by Elsevier Ltd. on behalf of Chinese Society of Aeronautics and Astronautics. This is an open access article under the CC BY-NC-ND license (http://creativecommons.org/licenses/by-nc-nd/4.0/).

Aluminum is an important energetic component of many solid

added to rocket propellants to increase the specific impulse

and raise the flame temperature.⁴ Thus, the research on alu-

minum has been an ongoing effort. Specifically, aluminum

has both high gravimetric calorific value (30.96 kJ/g) and vol-

umetric calorific value (83.59 kJ/cm³). Furthermore, due to its

wide availability, low cost, harmless formation, and non-toxic

propellants, explosives, and pyrotechnic formulations.^{1–3}

1. Introduction

20

21

22

23

24

25

26

27

It is

* Corresponding author. E-mail address: jzliu@zju.edu.cn (J. LIU).

Peer review under responsibility of Editorial Committee of CJA.

ELSEVIER Production and hosting by Elsevier

https://doi.org/10.1016/j.cja.2017.09.006

1000-9361 © 2017 Production and hosting by Elsevier Ltd. on behalf of Chinese Society of Aeronautics and Astronautics. This is an open access article under the CC BY-NC-ND license (http://creativecommons.org/licenses/by-nc-nd/4.0/).

Please cite this article in press as: ZHOU Y et al. Effect of particle size and oxygen content on ignition and combustion of aluminum particles, *Chin J Aeronaut* (2017), https://doi.org/10.1016/j.cja.2017.09.006

ARTICLE IN PRESS

86

87

88

89

90

91

92

93

94

95

96

97

98

99

100

101

102

103

129

2

28

29

characteristic, aluminum is also used as a cost-effective metal fuel.⁵

30 In its practical applications, aluminum is mixed with the 31 solid propellant in a certain proportion, and eventually releases the chemical energy in the form of combustion. So 32 the study of the ignition and combustion characteristics of alu-33 34 minum particles can guide the applications of aluminum based composite propellants and provide a theoretical basis for 35 studying the mechanism of aluminized composite propellant 36 combustion and improving the combustion efficiency of solid 37 propellants.6 38

39 The ignition and combustion characteristics of aluminum 40 and aluminum-based propellants have been studied over the past decades.^{7,8} Glassman and Brzustowski^{9,10} recognized that 41 aluminum combustion would be similar to combustion of dro-42 plets of a hydrocarbon fuel for which the D^2 law (D is the alu-43 minum particle size)is applicable for ignition and combustion, 44 depending on the melting and boiling points of the metal and 45 the oxide. Belyaev et al.¹¹ added a small amount of aluminum 46 powder into a solid propellant and found that the burning time 47 could be extended by increasing the aluminum particle size 48 from 10 μ m to 150 μ m. Bazyn et al.¹² employed the shock tube 49 to study the effects of atmospheric conditions on the burning 50 time, and found that the burning time is the shortest in O_2 , 51 and the longest in H₂O, and also shows a tendency to extend 52 with increase in the amount of oxidizer. Lynch et al.¹³ obtained 53 54 the same results as Bazyn et al. and found that the actual burning time was shorter than that predicted by the model of 55 diffusion-controlled combustion. Roberts et al.¹⁴ performed 56 a shock experiment under low pressure conditions, and found 57 that reducing the oxygen concentration caused extension of the 58 59 ignition time.

60 Until now, the ignition and combustion mechanisms of a single particle have not been studied thoroughly. Thus, the 61 62 research on the ignition and combustion behavior of small alu-63 minum particles is still an ongoing process. Particle size and 64 ambient atmosphere can significantly influence the ignition 65 and combustion characteristics of aluminum particles. However, very few studies have focused directly on the effects of 66 particle size and oxygen concentration. Most of the previous 67 68 reports have been on micron-sized aluminum larger than 10 µm 69 which is not applicable for practical engineering applications. It is necessary to separately study the nano-aluminum material 70 because the physical properties of nanocrystalline metals are 71 significantly different than those of bulk polycrystalline 72 metals.12 73

In this study, the aluminum microstructure was analyzed 74 using Scanning Electron Microscopy (SEM). The effects of 75 particle size and oxygen concentration on the ignition and 76 77 combustion characteristics of aluminum particles were systematically investigated using a laser ignition experimental system. 78 Lastly, employing a thermobalance, the effects of particle size 79 80 and oxygen concentration on the thermal oxidation character-81 istics were investigated.

82 **2.** Experimental and methods

83 2.1. Materials

The aluminum samples used in this study were obtained from Shanghai Shuitian Technology Co., Ltd., Shanghai, China. The nominal purity of the samples was 99%. The samples were dark grey at room temperature. The nominal size of the samples were 80 nm, 1 μ m, 10 μ m and 50 μ m, and the mean particle size were 102 nm, 3.53 μ m, 8.617 μ m and 37.75 μ m, respectively. As samples are of a non porous material, their specific areas are only affected by particle size.

2.2. Devices and methods

SEM images were taken by a Hitachi SU-70 field emission SEM, after gold sputtering for two minutes. Thermogravimetric (TG) analysis experiments on the samples were conducted on a TA-Q500 thermal analysis system. Approximately 2 mg of the samples was packed in Al_2O_3 crucible for each TG experiment. The samples were heated from room temperature to 1000 °C at a heating rate of 10 °C/min. The reaction gas was a mixture of nitrogen at a constant gas flow of 40 mL/min and oxygen at a flow of 10 mL/min, 27 mL/min, and 100 mL/min for different oxygen content experiments.

A laser ignition system (Fig. 1) was designed and con-104 structed for the laser ignition experiments. This experimental 105 setup has been widely used in metal ignition and combustion 106 studies.^{16–18} This system consisted of four parts, namely com-107 bustion diagnosis unit, laser ignition unit, gas regulation unit, 108 and data acquisition unit. The CO₂ laser was used to heat and 109 ignite the samples. An AvaSpec-3648 fiber optic spectrometer 110 was used to record the characteristic speactra of the samples 111 during the combustion stage in real time. A Redlake 112 GE4900-T12 color high speed camera, with a maximum reso-113 lution of 1024 dpi \times 768 dpi, recorded the changes of the flame 114 during combustion. The switch gear controlled the CO₂ laser 115 and fiber optic spectrometer at the same time. The flowmeter 116 and gas cylinder were set to control the atmosphere and gas 117 flow rate of the combustion chamber. In this experiment, the 118 laser power was set to 270 W, and firing time of the laser 119 was 1 s. Approximately 10 mg of the samples was loaded into 120 the Tungsten crucible in the combustion chamber. The reac-121 tion gas was a mixture of oxygen and nitrogen which contained 122 20%, 50%, 70%, and 100% oxygen, at a constant flow of 123 1 L/min. The combustion products were collected and 124 weighed. The burn-off rate was obtained by calculating the 125 amount of reacted aluminum from the weights of the combus-126 tion products. 127

3. Results and discussion 128

3.1. Microstructure analysis

Figs. 2 and 3 show the SEM images of the aluminum sample 130 with average particle size of 1 μ m, and its primary combustion 131 products after ignition by laser. Before ignition, the particles 132 have a spherical shape, and the size distribution is relatively 133 uniform, around 1 µm (Fig. 2). The particles were found to 134 be aggregated with each other, and the surface was not com-135 pletely smooth due to some protuberances and cavities. The 136 shape of the particles after ignition is irregular. There are 137 two distinctly different regions present in the sample (Fig. 3). 138 One region (Section A) consists of inhomogeneous spherical 139 particles clustered together, and the particles expanded in size 140 because of oxidation. The other region (Section B) is a section 141

Download English Version:

https://daneshyari.com/en/article/7153849

Download Persian Version:

https://daneshyari.com/article/7153849

Daneshyari.com