



# Thermal analysis of double-pipe heat exchanger in thermodynamic vent system



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## ABSTRACT

Full use and effective management of cold capacity are significant for improving the performance of heat exchanger in the thermodynamic vent system (TVS). To understand the operation principle of TVS easily, the thermodynamic analysis, based on the ideal gas state equation and energy conservation equation, is detailed introduced. Some key operation parameters are optimized and suggested. As the low mass flow rate and low heat fluxes are involved in flow boiling of the annular pipe fluid, the Kandlikar's boiling heat transfer correlation is selected to predict the flow boiling process, after validated with the related experimental results. One quasi-steady state model is established to investigate the heat transfer performance of double-pipe heat exchanger in normal gravity, with the bulk fluid natural convection, annular pipe two-phase boiling and inner pipe forced convection coupled from outside to inside. Determined by the local pressure and temperature, the fluid thermophysical properties are variable with the pipe length and time. With the variable fluid thermophysical properties, both the static analysis and the transient thermal performance of TVS heat exchanger are investigated respectively. Meanwhile, effects of the external natural convection and the pipe sizes on the thermal and flow performance of heat exchanger are detailed researched and analyzed. Some valuable conclusions are obtained and significant to optimize the TVS heat exchanger design.

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## 1. Introduction

With the inevitable space heat leakage, cryogenic fuel tank pressure will increase rapidly, which is dangerous for tank safety operation [1]. As long duration storage of cryogenics in space is greatly important for future exploration mission, some effective pressure control measures have to be conducted. Generally, direct venting is used for relieving the fuel tank pressure for short term storage. While it is applied to long time orbit storage, for instance, several weeks or several months, more propellant boil-losses cost and launch weight will also increase. Therefore, to effectively maintain tank pressure and reduce the propellant venting losses, active pressure control measure must be adopted. By comparisons and verifications, TVS is proven to have a promising application prospect [2,3], with validly maintaining tank pressure through active-venting without resettling.

In the past 20 years, NASA has conducted many theoretical analysis, experimental studies and numerical simulations on TVS.

In 1994, Nguyen [4] developed a performance prediction program to investigate the zero-g TVS with different heat transfer modes considered. Based on Nguyen's computational program, Hastings et al. [5] investigated the performance of spray bar TVS for on-orbit LH<sub>2</sub> tank with theoretical analysis. Afterwards, TVS experiments related to LH<sub>2</sub> [6,7], LN<sub>2</sub> [8] and LCH<sub>4</sub> [9] were conducted successively on the multipurpose hydrogen test bed (MHTB) of Marshall Space Flight Center. As different liquid filling heights, heat fluxes, tank pressure control bands, pressurization gases and TVS operation models were detailed experimentally investigated in the ground condition, the tank pressure control has been proven to be better realized by TVS. During the 2008–2010, Ho and Rahman [10,11] have conducted some numerical calculations to investigate the performance of zero boil-off tank by establishing 2D and 3D calculation model. By combining a heat pipe and spray nozzle used in the ZBO tank, it turns out that the tank pressure has been effectively controlled. Recently, Kartuzova et al. [12] presented a CFD spray model to predict the spray flow process with Euler-Lagrange approach to track the spray droplets. Meanwhile, the particle tracking is performed by coupling the spray model with VOF model, with considering the interface heat and mass transportation.

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