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## Fracture resistance of aluminum multilayer composites

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#### ABSTRACT

The so-called material inhomogeneity effect, which influences the crack driving force in inhomogeneous materials, has recently been found to contribute to the excellent behavior of some fracture resistant biomaterials. The current study aims to transfer these findings to engineering materials, by experimentally investigating the fracture behavior of multilayers consisting of a high-strength aluminum alloy and thin, soft polymer interlayers. Structures in crack divider and crack arrester (CA) configuration are tested. The results show that the structures in CA configuration exhibit a tremendously improved fracture resistance compared to the homogeneous bulk material. The reason is that cracks are completely arrested in the soft interlayers. This effect appears without delamination, i.e. it is basically different from the well-known delamination effect on weak interlayers.

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#### 1. Introduction

The heat-treatable Al–Zn–Mg–Cu (7xxx) alloys have the highest strength to weight ratio of all aluminum alloys, making them ideal candidates for lightweight constructions. Nevertheless, the application of these alloys in safety critical structures is limited due to their rather poor resistance to fracture and fatigue.

To overcome the dilemma that increasing strength leads to a toughness decrease in many materials and alloys, damage tolerant multilayered systems have been invented and intensively investigated in the past decades [1]. The basic idea behind these multilayered structures is to increase the fracture resistance by the delamination of weak interfaces, while at the same time preserving the high material strength [1–4]. Two configurations, denominated as crack divider (CD) and crack arrester (CA) configuration, are examined in these publications. In the CD configuration the crack plane is normal to the layers and the crack propagates through all layers at once (Fig. 1a), while in the CA configuration the crack also propagates perpendicular to, yet sequentially through the layers (Fig. 1b).

In the first works on the subject, metal/metal multilayer systems with weak interlayers, introduced by soldering, explosion-cladding, or roll-bonding, were investigated with respect to their impact toughness [3] or fracture toughness [4]. Compared to a homogeneous material of the same total thickness, the fracture behavior was improved in both configurations. This fact has been attributed to the delamination of the weak interfaces in the multilayers. The delamination diminishes the hydrostatic stress state around the crack tip, which leads to a retardation of void-initiation [5] and a reduction of the void growth rates [6] and, therefore, to an increased fracture toughness. Additionally, in the CA configuration the crack tip is blunted when it grows into the delaminated area, and a re-initiation of the crack is necessary for further crack growth.

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Nomenclature	
a	crack length
а а.	initial crack longth
$\Delta a$	crack extension
Δι	plactic area under the load vs. load line displacement surve
A <sub>pl</sub>	plastic area under the load vs. load-inte-displacement curve
A <sub>el</sub>	total area under the load up load line displacement surve
A <sub>tot</sub>	ligen act length
D	ingament length
D <sub>arr</sub>	igament length after crack arrest
В	specimen thickness
C <sub>inh</sub>	material innomogeneity term
E	Young's modulus
F	applied load
F <sub>max</sub>	maximum load reached in the fracture mechanics test
H	gauge length
J	J-integral
Jel	elastic component of the J-integral
$J_{\rm pl}$	plastic component of the J-integral
J <sub>tip</sub>	near-tip J-integral
Jfar	far-field J-integral
J <sub>max</sub>	maximum J-integral
$J_{F_{\max}}$	J-integral value at $F_{\rm max}$
Jic	fracture toughness in terms of the J-integral
Jc	fracture initiation toughness in the CA configuration
K	stress intensity factor
K <sub>JC</sub>	critical stress intensity factor calculated from $J_{IC}$
L	length plastically deformed in the fracture mechanics specimen
n	strain hardening exponent
$N_{u}$	number of undamaged layers
$r_{y}^{(red)}$	plastic zone radius in plane stress
R	crack growth resistance
$R_{p0.2}$	yield strength
<i>K</i> <sub>m</sub>	ultimate tensile strength
t	layer thickness
t <sub>opt</sub>	optimum layer thickness
$v_{LL}$	load-line-displacement curve
V	volume plastically deformed in the fracture mechanics specimen
VV	specifien width
γ Γ	nondimensional factor to account for crack growth effects on J
1	surface energy needed to create new fracture surfaces
3*	engineering strain
3	
$\varepsilon_{\rm F}$	engineering fracture strain
U3	uniform elongation
$\eta^{*}$	nondimensional factor connecting $A_{pl}$ to $f_{pl}$
η	Decisional factor connecting A <sub>tot</sub> to J
V	
0	engineering suess
o <sub>max</sub>	maximum nonmial stress in the figament after Clack affest
	crack divider configuration
	crack tip opening displacement
	compact specimen
C(T)	single edge notch tension specimen
JL(I)	single edge noten tension specifien

Similar observations have been made in laminates where metal layers were combined with polymer adhesives [7,8]. A recent development of these polymer/metal multilayers are fiber metal laminates, where the polymer adhesives are reinforced by glass- or aramid fibers [9,10]. The fibers bridge the zone behind the crack tip if the crack grows into the polymer, which should lead to a further improvement of the fracture behavior of the laminate.

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