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Study of crack orientation and fatigue life prediction in biaxial fatigue with critical plane models



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1. Introduction

ABSTRACT

The crack initiation and propagation behaviour of St52-3N steel is investigated under biaxial conditions. Short crack behaviour is studied and pearlite bands are found to modify the crack growth rate. The behaviour under proportional and non-proportional (90°) loading is studied with different critical plane models; Wang–Brown, Fatemi–Socie, Smith–Watson– Topper, Liu I and Liu II. Smith–Watson–Topper and Liu I are found to best predict the crack orientation under proportional loading and Fatemi–Socie and Liu II under non–proportional loading. The best predictions in terms of fatigue life are obtained by Fatemi–Socie, both under proportional and non–proportional loads.

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The majority of mechanical components experience multiaxial cyclic loading conditions. As a consequence two or three principal stresses and strains change in time, both in magnitude and direction. A number of theories have been proposed to estimate fatigue life under such complex conditions [1–4]. Of these theories, critical plane approaches were found to be very effective because they are based on the fracture mode or the initiation mechanism of cracks. They are founded on the physical observations that cracks initiate and grow on preferred planes. Critical plane approaches proved to be very successful in a wide range of applications and materials, including high strength 42CrMo steel [5], variable amplitude on a car component 1045 steel [6], S45C steel under variable amplitude loading [7], SNCM630 steel [8], 1045HR and 304 stainless steels [9]. Moreover critical plane approaches can predict not only the fatigue life under multiaxial loading, but also the cracking orientation. This latter element is of key importance since it influences, together with the geometric discontinuities and the multiaxial loading, the crack path during the propagation stage. The crack path in critical components can determine whether failure is catastrophic or not. In addition, accurate information about cracking direction is very helpful for selecting the most convenient non-destructive testing procedures. Nevertheless poor agreement is often observed between theoretically predicted and experimental tests [10]. As a consequence, since 2003 a series of international conferences have been devoted specifically to the path taken by cracks [11]. Even though critical plane approaches are able to provide information about fatigue lives and crack orientation, most works focus on fatigue life, with little attention paid to cracking orientation

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6	fatigue strength exponent
bγ	fatigue strength exponent
	fatigue ductility exponent
Y	fatigue ductility exponent
la/dN	crack growth rate
:	parameter represents the influence of the normal stress in Fatemi-Socie model
7	cyclic strength coefficient
ζ'_{γ}	cyclic strength coefficient
ı′ [`]	cyclic hardening exponent
ι 'γ	cyclic hardening exponent
V_f	fatigue life, number of cycles
5	sensitivity factor for Brown–Miller model
'a	angular strain amplitude
'f	fatigue ductility coefficient
$\Delta \gamma_{\rm max}/2$	maximum angular strain amplitude
$\Delta \varepsilon_n$	range of normal strain acting on the plane subjected to $\Delta \gamma_{max}/2$
$\Delta W_{\rm I}$	axial work
ΔW_{II}	shear work
а	axial strain amplitude
eq	von Mises equivalent strain
f	fatigue ductility coefficient
)	angle between longitudinal direction of the specimen and the normal direction
'e	Poisson's ration in elastic deformation
'p	Poisson's ration in plastic deformation
σ_a	axial stress amplitude
rcr	critical buckling stress
$\sigma_{n,\max}$	maximum normal stress acting on the plane subjected to $\Delta \gamma_{max}/2$
\overline{r}_{x}	normal stress
y /	yield strength
Гу _/	cyclic yield strength
\mathcal{F}_{f}	Idligue strength coefficient
'u	unimate tensite stress
a	angular succes amplitude
'xy -/	cuclic yield strength
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[2,5–8,12,13]. Among the few works that have paid attention to crack orientation, Reis and co-authors used critical plane models to predict the crack orientation during early crack growth [14] and to predict fatigue life [15] for a wide range of axial/torsional loading paths. Other researchers employed an innovative approach by combining principal stress field and finite element method to predict the crack orientation under proportional [16] and both proportional and non-proportional loading [17]. As a consequence, one of the key aims of this work is to study the crack orientation under different load combinations.

There are a number of issues on multiaxial fatigue in which researchers do not completely agree [18]. For example, one of the most popular criterion [19] was found both to overestimate [20,21] and underestimate [8] fatigue life. Accordingly a comparative study of five widely used critical plane criteria (Wang–Brown [19], Fatemi–Socie [22] and Smith–Watson–Topper [23], Liu I and Liu II [24]) is shown here.

Biaxial experimental tests are difficult, expensive and time-consuming, hence the amount of biaxial data available in the literature is not as large as for uniaxial fatigue [9,25]. The current work aims to characterise the fatigue behaviour under biaxial conditions both during initiation and propagation stages. Since both stages depend on very different mechanisms [26,27], different methodologies will be used to study them. First, a methodology for characterising the crack growth of short fatigue cracks under biaxial loading is presented. Then a series of biaxial tests conducted to evaluate in-phase and out-of-phase conditions are described. Detailed observations on the cracking orientation are made for the initiation and propagation stages. Comparison between several critical plane theories is performed both in terms of fatigue life and crack orientation. Each section includes a discussion of the results shown in that specific section. An overall short discussion is included at the end. Practical issues often neglected in publications, such as how and where the orientation is measured and its length-scale are also discussed.

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