Accepted Manuscript

Experimental investigation on the effect of swirling flow on combustion characteristics and performance of solid fuel ramjet

Omer Musa, Li Weixuan, Chen Xiong, Gong Lunkun, Liao Wen-He

PII: S0094-5765(18)30261-3

DOI: 10.1016/j.actaastro.2018.04.055

Reference: AA 6855

To appear in: Acta Astronautica

Received Date: 5 February 2018

Revised Date: 24 April 2018 Accepted Date: 30 April 2018

Please cite this article as: O. Musa, L. Weixuan, C. Xiong, G. Lunkun, L. Wen-He, Experimental investigation on the effect of swirling flow on combustion characteristics and performance of solid fuel ramjet, *Acta Astronautica* (2018), doi: 10.1016/j.actaastro.2018.04.055.

This is a PDF file of an unedited manuscript that has been accepted for publication. As a service to our customers we are providing this early version of the manuscript. The manuscript will undergo copyediting, typesetting, and review of the resulting proof before it is published in its final form. Please note that during the production process errors may be discovered which could affect the content, and all legal disclaimers that apply to the journal pertain.



ACCEPTED MANUSCRIPT

Experimental investigation on the effect of swirling flow on combustion characteristics and performance of solid fuel ramjet

Omer Musa^{a,b,*}, Li Weixuan^a, Chen Xiong^a, Gong Lunkun^a, Liao Wen-He^a

^aSchool of Mechanical Engineering, Nanjing University of Science and Technology, Nanjing 210094, P R China ^bMechanical Engineering Department, Omdurman Islamic University, Omdurman, Sudan

Abstract

Solid-fuel ramjet converts thermal energy of combustion products to a forward thrust without using any moving parts. Normally, it uses air intake system to compress the incoming air without swirler. A new design of swirler has been proposed and used in the current work. In this paper, a series of firing tests have been carried out to investigate the impact of using swirl flow on regression rate, combustion characteristics, and performance of solid-fuel ramjet engines. The influences of swirl intensity, solid fuel port diameter, and combustor length were studied and varied independently. A new technique for determining the time and space averaged regression rate of high-density polyethylene solid fuel surface after experiments has been proposed based on the laser scan technique. A code has been developed to reconstruct the data from the scanner and then used to obtain the three-dimensional distribution of the regression rate. It is shown that increasing swirl number increases regression rate, thrust, and characteristic velocity, and, decreases air-fuel ratio, corner recirculation zone length, and specific impulse. Using swirl flow enhances the flame stability meanwhile negatively affected on ignition process and specific impulse. Although a significant reduction of combustion chamber length can be achieved when swirl flow is used. Power fitting correlation for average regression rate was developed taking into account the influence of swirl number. Furthermore, varying port diameter and combustor length were found to have influences on regression rate, combustion characteristics and performance of solid-fuel ramjet.

Keywords:

Swirling flow; Solid-fuel ramjet; Combustion; Solid fuel; Energy; Propulsion.

1. Introduction

- Ramjet or flying stovepipe is an air-breathing propulsion engine that contains no moving parts in which
- the flight is based on ram air compression thereby fixed components are used to compress and accelerate
- 4 intake air using ram effect. Then, the thermal energy is imparted to the gas from burning fuel (solid, liquid,

Email address: omer.musa1@hotmail.com; Tel.: +8613951981784; fax: +862584315931 (Omer Musa)

^{*}Corresponding author.

Download English Version:

https://daneshyari.com/en/article/8055497

Download Persian Version:

https://daneshyari.com/article/8055497

<u>Daneshyari.com</u>