## Accepted Manuscript

Overload control of artificial gravity facility using spinning tether system for high eccentricity transfer orbits

Xing-wang Gou, Ai-jun Li, Hao-chang Tian, Chang-qing Wang, Hong-shi Lu

PII: S0094-5765(17)31368-1

DOI: 10.1016/j.actaastro.2018.03.005

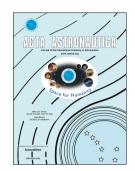
Reference: AA 6747

To appear in: Acta Astronautica

- Received Date: 10 October 2017
- Revised Date: 13 February 2018
- Accepted Date: 2 March 2018

Please cite this article as: X.-w. Gou, A.-j. Li, H.-c. Tian, C.-q. Wang, H.-s. Lu, Overload control of artificial gravity facility using spinning tether system for high eccentricity transfer orbits, *Acta Astronautica* (2018), doi: 10.1016/j.actaastro.2018.03.005.

This is a PDF file of an unedited manuscript that has been accepted for publication. As a service to our customers we are providing this early version of the manuscript. The manuscript will undergo copyediting, typesetting, and review of the resulting proof before it is published in its final form. Please note that during the production process errors may be discovered which could affect the content, and all legal disclaimers that apply to the journal pertain.



## Overload control of artificial gravity facility using spinning tether system for high eccentricity transfer orbits

Xing-wang Gou<sup>1</sup>, Ai-jun Li<sup>1</sup>, Hao-chang Tian<sup>1</sup>, Chang-qing Wang<sup>\*1</sup>, and Hong-shi Lu<sup>1</sup>

1. School of Automation, Northwestern Polytechnical University, Xi'an Shaanxi 710072, China

Abstract: As the major part of space life supporting systems, artificial gravity requires further study before it becomes mature. Spinning tether system is a good alternative solution to provide artificial gravity for the whole spacecraft other than additional devices, and its longer tether length could significantly reduce spinning velocity and thus enhance comfortability. An approximated overload-based feedback method is proposed to provide estimated spinning velocity signals for controller, so that gravity level could be accurately controlled without complicated GPS modules. System behavior in high eccentricity transfer orbits is also studied to give a complete knowledge of the spinning stabilities. The application range of the proposed method is studied in various orbit cases and spinning velocities, indicating that it is accurate and reliable for most of the mission phases especially for the final constant gravity level phase. In order to provide stable gravity level for transfer orbit missions, a sliding mode controller based on estimated angular signals is designed for closed-loop control. Numerical results indicate that the combination of overload-based feedback and sliding mode controller could satisfy most of the long-term artificial gravity missions. It is capable of forming flexible gravity environment in relatively good accuracy even in the lowest possible orbital radiuses and high eccentricity orbits of crewed space missions. The proposed scheme provides an effective tether solution for the artificial gravity construction in interstellar travel.

<sup>\*</sup> Corresponding author. E-mail address: wangcq@nwpu.edu.cn (Chang-qing Wang).

Download English Version:

## https://daneshyari.com/en/article/8055556

Download Persian Version:

https://daneshyari.com/article/8055556

Daneshyari.com