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Fuzzy Adaptive Non-affine Attitude Tracking Control for a Generic Hypersonic Flight Vehicle

Yuhui Wang¹, Mou Chen¹, Qingxian Wu¹, Jun Zhang²

 Department of Automation Engineering, Nanjing University of Aeronautics and Astronautics, Nanjing, 211106, China
 Science and Technology on Space Physics Laboratory, Beijing Institute of Nearspace

: Science and Technology on Space Physics Laboratory, Beijing Institute of Nearspace Vehicle's Systems Engineering, Beijing, 100076, China

Abstract

A fuzzy adaptive non-affine attitude tracking control method is proposed for a generic hypersonic flight vehicle (HFV). Due to the complexities of the hypersonic flows and nonlinear dynamics, the aerodynamic coefficients of the HFV are dependent not only on the attack of angle, sideslip angle, angular rates, and Mach number, but also on the deflection angles of the control surfaces. This cause the attitude tracking control problem becomes a non-affine multi-input-multi-output (MIMO) one. By analyzing the characteristics of the aerodynamic coefficients, it can be found that the non-affine terms have a great influence on the attitude tracking controller is designed using fuzzy sliding mode adaptive techniques with consideration of the non-affine nonlinear aerodynamic coefficients. The proposed controller has good practicability because it does not need to know the exact bound values of the uncertain-

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