Accepted Manuscript

Numerical study of an airfoil with riblets installed based on large eddy simulation

Yufei Zhang, Haixin Chen, Song Fu, Wenjiao Dong

 PII:
 S1270-9638(17)32013-8

 DOI:
 https://doi.org/10.1016/j.ast.2018.05.013

 Reference:
 AESCTE 4566

To appear in: Aerospace Science and Technology

Received date:3 November 2017Revised date:18 March 2018Accepted date:12 May 2018



Please cite this article in press as: Y. Zhang et al., Numerical study of an airfoil with riblets installed based on large eddy simulation, *Aerosp. Sci. Technol.* (2018), https://doi.org/10.1016/j.ast.2018.05.013

This is a PDF file of an unedited manuscript that has been accepted for publication. As a service to our customers we are providing this early version of the manuscript. The manuscript will undergo copyediting, typesetting, and review of the resulting proof before it is published in its final form. Please note that during the production process errors may be discovered which could affect the content, and all legal disclaimers that apply to the journal pertain.

Numerical Study of an Airfoil with Riblets Installed

Based on Large Eddy Simulation*

Yufei Zhang[†], Haixin Chen[‡], Song Fu[§], Wenjiao Dong^{**} (School of Aerospace Engineering, Tsinghua University, Beijing, 100084, China)

Abstract:

The implicit large eddy simulation method is applied in a riblet drag reduction study of a lowspeed airfoil. The numerical method is tested by a channel flow with streamwise riblets installed on one side. The mean velocity profile, root mean square of velocity and Reynolds shear stress match well with reference direct numerical simulation data. Next, numerical investigation is conducted on an airfoil called Eppler E374 at free stream Mach number 0.2, angle of attack 3° and Reynolds number 2.0×10^5 . The simulated flow around the airfoil without or with a numerical trip is in good agreement with the experimental data. When isosceles triangle riblets are installed on the airfoil along the streamwise direction, the lift coefficient is increased, and the friction drag is decreased. The pressure distribution on the airfoil is slightly changed, corresponding to the increased pressure drag. The Reynolds stresses are greatly reduced by the riblets at locations of strong pressure gradient. The result of the power spectrum density of pressure shows that the high-frequency fluctuations are suppressed when the riblet film is installed on the airfoil, and the vortex structures in the boundary layer are also reduced.

Keywords:

large eddy simulation, streamwise riblet, airfoil, drag reduction, numerical trip

1. Introduction

Aerodynamic drag includes friction drag, induced drag and wave drag. Friction drag caused by turbulent flow plays an important role, accounting for approximately 50% of the total drag of a civil aircraft. Because the near-wall turbulent coherence structure is relevant to the high-friction drag of the boundary layer, flow control methods of suppressing the turbulence structure are usually employed to reduce the friction drag. The use of riblets is a passive flow control technique of friction drag reduction that restricts the streamwise vortex and suppresses the generation of hairpin vortex in the boundary layer; consequently, this technique reduces the Reynolds stress and decreases the momentum and energy loss inside the boundary layer. Many investigators are interested in the riblet technique because of its efficiency and implementability.

Experimental study is a common and convenient method to test the drag reduction effect of riblets. Vukoslavcevic et al. [1] studied the effect of riblets at $Re_0=10^3$ and found that the most efficient region for depressing the turbulent fluctuation is under 4% of the boundary layer height.

National Natural Science Foundation of China (11572177).

This work was supported by the National Key Basic Research Program of China (2014CB744801) and the

[†] Corresponding author, associate professor; Email: zhangyufei@tsinghua.edu.cn

⁺ Professor; Email: chenhaixin@tsinghua.edu.cn

[§] Professor; Email: fs-dem@tsinghua.edu.cn

^{**} Master student; Email: dwj1826@163.com

Download English Version:

https://daneshyari.com/en/article/8057581

Download Persian Version:

https://daneshyari.com/article/8057581

Daneshyari.com