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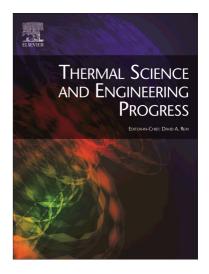
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Nomenclature

A = area, cm^2

 A_2 = diffuser (engine inlet) exit area, cm²; combustor entrance area

 A_4 = combustor exit area, cm²

 A^*/A = throat area to combustor area ratio, [Eq. (41)]

 A^* = throat area, cm²

a_o = freestream speed of sound, m/s

a₉ = speed of sound at nozzle exit, m/s

B = defined in Eq. (10b)

C = defined in Eq. (14)

 c_p = specific heat at constant pressure, kJ/(kg K)

F = thrust, N

 F/\dot{m}_0 = specific thrust, N/(kg/s)

 F/A_2 = thrust flux, N/cm²

f = mass based fuel-to-air ratio

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